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## 1 Single mode Lamb wave excitation at high frequency-thickness products using a 2 conventional linear array transducer

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## 9

10 Abstract

11 Lamb wave excitation at high frequency-thickness products offers a potential solution for high-resolution guided wave testing. The method is attractive for crack imaging and corrosion mapping, especially in hidden 12 locations where direct access is limited. However, multiple modes may propagate, complicating signal 13 14 interpretation, which is undesirable. In this work, a systematic approach is presented, in an effort to determine 15 the influence of the key parameters related to single higher order Lamb wave mode excitation with a 16 conventional linear array transducer. Specifically, a linear time delay law is used to enhance a targeted mode, 17 while the array's length, pitch and apodisation profile remain to be optimally selected. First, an analytical 18 solution is derived based on modal analysis. This provides a natural decomposition of the amplitude of a guided 19 wave mode to the product of the response of a single element and the excitation spectrum, which is related to 20 properties of the array. Then, a key observation is made, associating the excitation spectrum to the directivity 21 function for bulk wave phased array steering. This allows the application of well established phased array 22 analysis tools to guided wave phased array excitation. In light of this fact, minimisation of the spectrum's 23 bandwidth, elimination of the grating lobes and derivation of an apodisation profile are performed, to enhance 24 the purity of the targeted mode. Finally, experiments conducted on an aluminium plate verify the above 25 theoretical results. The Full Matrix is acquired, and all signals are reconstructed synthetically.

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Keywords: Lamb waves, Higher order mode region, Single mode excitation, Unidirectional propagation,
Apodisation

## 30 1. Introduction

31 Ultrasonic guided waves (UGW) are extensively used in structural health monitoring and non-destructive 32 evaluation applications. They are commonly employed for the inspection of a large variety of structural assets, 33 such as pipelines, storage tanks and pressure vessels. In this work, attention is shifted to a special class of UGW, 34 namely Lamb waves, propagating in plates or plate-like waveguides. Lamb waves decompose to symmetric or 35 antisymmetric modes. They can be classified based on their frequency-thickness product [1,2], which 36 determines their key properties, such as phase velocity, dispersion and modal density. The region below 10 37 MHz·mm has been extensively investigated by several authors, particularly for steel and aluminium samples 38 [3–15]. Recently, there has been a research interest in the region above 15 MHz·mm [1,2,16–22], called here 39 higher order mode region, which is very attractive, because of the potential for medium-range high-resolution 40 inspection of relatively thick samples at high frequencies. For example, operation at 20 MHz·mm implies guided waves can be generated on a 10 mm thick plate with a 2 MHz probe. Potential applications include the 41 in-service inspection of the annular plate of storage tanks [1], thickness gauging [20], crack imaging [23,24] 42 43 and inspection for corrosion on pipe networks, especially in hidden locations, such as corrosion under piper 44 supports [21,22].

Lamb waves are commonly excited with piezoelectric, laser or electromagnetic acoustic transducers [25]. Among others, piezoelectric transducer excitation includes wedge [18], comb [26], interdigital [13], periodic [27], phased comb [8,28] or apodised phased comb [29]. The latter is an extension of phased comb excitation, - 2 -

48 describing the case where both time delays and voltage amplitudes vary across the elements of a conventional 49 1-D linear array probe with individually addressable elements. Such arrays are commonly deployed for bulk 50 phased array ultrasonic inspections due to their flexibility and therefore guided wave excitation utilising these probes is advantageous, as they are readily available and mature. However, transducer size constrains and 51 52 scanning resolution requirements for common engineering structures obligate manufacturing usually above 1 53 MHz. Commonly, operation at low frequency thickness products is preferred [27,30,31], below 3.5 MHz mm, 54 as it is easier to target a single mode. However, this requires thin waveguides, approximately up to 3.5 mm 55 thick. Recent work has shown potential in exciting Lamb waves around 10 MHz mm [8,29,32]. Veit and 56 Bélanger [8] reported the generation of a single Lamb wave mode using a 64-element phased array probe centred 57 at 1.5 MHz on a 5 mm thick aluminium sample. They used a linear time delay law and uniform voltage 58 amplitudes across the array elements. Cirtautas et al. [29] demonstrated the feasibility of generation of the 59 symmetric mode S3 on a 10 mm sample with a linear 1 MHz array mounted on an angled wedge using a two-60 sided excitation approach. The interaction of S3 with corrosion-like defects was simulated, showing the potential of the mode for corrosion defect detection and classification. 61

62 Guided wave phased array excitation is influenced by two key objects, namely the excitation and frequency 63 spectrum [7]. The frequency spectrum is determined by the temporal profile of the applied pressure load. This 64 profile is defined as the convolution of the excitation signal and the impulse response of the array elements. In guided wave applications, the excitation signal is typically a finite-cycled toneburst, controlled by the centre 65 frequency, number of cycles and applied window [15]. The impulse response of an element of the array depends 66 67 on the specific array design and material properties. Although this response might vary across the elements of 68 the array, these variations are usually small and thus all elements are considered identical. Ultimately, the 69 frequency spectrum is a bandpass filter, characterised by its centre frequency and bandwidth [33]. To excite a 70 single mode, a narrowband spectrum is preferred, as unwanted modes outside the band range are filtered out. 71 The excitation spectrum depends on the number of elements, pitch, time delay law and apodisation profile 72 employed. In the frequency-wavenumber domain, assuming linear time delays and uniform voltage amplitudes, 73 the points (f, k) where the spectrum maximises appear as straight lines (excitation beams). Their slope is 74 determined by the pitch and the applied linear time delay law. Among the infinitely many excitation beams, 75 one passes through the origin of the frequency-wavenumber domain. In the frequency-phase velocity domain, 76 this beam appears as a straight horizontal line at constant phase velocity. Commonly, the time delays are 77 adjusted so that the same excitation beam crosses the desired wave mode at the centre frequency. This way the 78 two spectra intersect on top of the targeted mode, which is then generated and propagates in the waveguide.

79 In this work, emphasis is placed on guided wave excitation at the higher order mode region, using a 80 conventional linear array. Employing a linear time delay law, a single low dispersion higher order mode is targeted at 20 MHz·mm. For the first time, a systematic analysis for guided wave excitation is presented where 81 82 the length, pitch and apodisation profile are appropriately determined to optimise single mode excitation. More specifically, the influence of the number of elements on the bandwidth of the excitation spectrum is investigated. 83 84 The effect of the pitch on the elimination of the grating lobes is studied in detail. The possibility of enhancing 85 the purity of the targeted mode using an apodisation profile is explored. Moreover, the targeted guided wave mode propagation is unidirectional. This is due to satisfaction of a condition involving the pitch and wavelength 86 87 of the selected mode. All conditions presented in this work are derived analytically and verified experimentally.

The organization of this paper is as follows. First, in Section 2, a model based on modal analysis is derived. The array is modelled as a transient periodic pressure load acting on the top surface of the plate, leading to an explicit expression between the amplitudes of any mode and the parameters relevant to the excitation. Then, a set of conditions related to the number of elements, pitch and apodisation profile is provided in Section 3. In Section 4, experimental results are presented, using a dataset acquired with the Full Matrix Capture (FMC) method. Finally, in Section 5, key conclusions are drawn.

- 3 -

#### 95 2. Analytical solution for the apodised phased comb array excitation problem based on modal analysis

96 Guided wave excitation studies are commonly based on a suitable analytical method. Among others, the 97 normal mode expansion method [34,35] and methods based on integral transforms, such as the Fourier transform 98 [7] are commonly utilised. This work presents an alternative technique based on modal analysis, to solve the 99 apodised phased comb array excitation problem. Besides its natural simplicity, the method offers some 100 advantages. Specifically, it treats transient loads directly, without having to express the input signal as a 101 superposition of time harmonic excitation sources using a Fourier transform [34]. Furthermore, there is no need 102 to define new concepts, such as a new orthogonality condition [34]. Throughout the analysis, all quantities 103 remain finite.



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**Figure 1.** Apodised phased comb array excitation model. The plate has finite length  $\ell$  and thickness *d*. The pressure load  $p_r$  on the top surface of the plate models the r<sup>th</sup> element. A local frame (x', z') is used to express the solution of the r<sup>th</sup> element. The global frame (x, z) coincides with the local frame of the first element.

108 Consider a homogeneous isotropic plate of length  $\ell$ , thickness d = 2h and width b. The equation of motion 109 after omitting external loads appears in the form [7]

$$(1) (1) \nabla (\nabla \cdot \underline{u}) + \mu \nabla^2 \underline{u} = \rho \underline{\ddot{u}},$$

111 where  $\lambda, \mu$  are the Lamé constants and  $\rho$  is the density of the plate. Assuming plane strain condition, the 112 displacement field with respect to a right-handed frame (x, y, z) located at the centre of the plate appears in the 113 form  $\underline{u} = (u^1, 0, u^3)$ . The cross section of the plate at y = b/2 is shown in Figure 1. Separation of variables in 114 space and time

 $u^{1} = X^{1}(x) X^{1}(z) T(t), \qquad u^{3} = X^{3}(x) X^{3}(z) T(t)$ (2)

and after substituting (2) in (1) and employing the free-free boundary conditions on the top and bottom 116 surfaces of the plate fully determines the through thickness profiles  $X^{1}(z)$  and  $X^{3}(z)$  of the modes in the axial 117 118 and thickness direction, respectively [7]. Similarly, the modal profiles along the axial direction,  $X^{1}(x)$  and 119  $X^{3}(x)$ , are determined by the boundary conditions on the left and right boundaries. However, in a guided wave 120 excitation study, waves are assumed to propagate far from the left and right boundaries; thus, a particular choice 121 of boundary conditions does not affect wave propagation. Therefore, the boundary conditions at the left and 122 right edges can be arbitrarily chosen. In fact, any complete orthogonal set of functions in  $L^2([-\ell/2, \ell/2])$  can be selected. In this work, the set  $\{1, \cos(kx), \sin(kx) \mid k = \frac{2\pi\alpha}{\ell}, \alpha = 1, 2, 3...\}$  is employed. As shown in [36], 123 'if  $X^{1}(x)$  is a sine function,  $X^{3}(x)$  must be a cosine function, and vice versa'. Note that for simplicity in 124 125 notation, X is used to describe both the through thickness and axial displacement profiles, which are 126 distinguished from the argument. The total solution can then be expanded in an infinite series form [37],

127  $\underline{u} = \tau^i \underline{\hat{X}}_i, i = 1, 2, \dots, \infty$ 

128 where  $\underline{\hat{X}}_i$  is the *i*<sup>th</sup> normalised eigenfunction such that  $\underline{\hat{X}}_i = \frac{1}{\sqrt{g_{ii}}} \underline{X}_i$  and  $g_{ij}$  is the metric tensor [38] defined as 129  $g(\underline{X}_i, \underline{X}_j) = g_{ij} = b \int_{-h}^{h} \int_{-\ell/2}^{\ell/2} \rho \underline{X}_i \underline{X}_j dx dz.$ 

130 The time dependent coefficients  $\tau^i$  are determined according to [37]

131 
$$\ddot{\tau}^{i} + \omega_{i}^{2} \tau^{i} = f_{i}, f_{i} = \int_{-l/2}^{l/2} p(x,t) \, \hat{X}_{i}^{3}(x) \, \hat{X}_{i}^{3}(z=h) \, dx \,, \tag{3}$$

132 where the eigenfrequency  $\omega_i$  related to mode  $\hat{X}_i$  is determined from the dispersion equations and

133 
$$p(x,t) = \sum_{r=0}^{N-1} p_r$$

is the excitation load modelling an array with N elements, width w and pitch s, as shown in Figure 1. The size of the elements in the y-direction is assumed to be much larger than their width (x-direction); therefore, the plane strain condition holds [7]. In what follows, emphasis is placed on deriving an analytical expression for the  $u^3$  component of the displacement field at the top surface of the plate. This is meaningful, as this is exactly the component sampled by a linear array. For simplicity in the notation,  $u^3(x, h, t)$  is denoted as u(x, t). The expression for  $u^1$  and other fields, such as velocity or stress fields, is similar.

140 The solution for an arbitrary element r of the array is expressed in the local frame x' = x - rs,  $t' = t - t_r$ , as shown again in Figure 1. The excitation load is given by

$$p_{\rm r}(x',t') = A_{\rm r}g(x';w)h(t';M),$$
(4)

143 where  $A_r$  and  $t_r = r\tau$  are the maximum voltage amplitude and linear time delay applied to the  $r^{\text{th}}$  element, 144 respectively, h(t') is an *M*-cycle normalised toneburst with centre frequency  $f_e$  and g(x';w) is the normalised 145 pressure distribution which is considered identical for all elements and is usually modelled as piston-source, 146 parabolic-source distribution or a similar window function of width w. Substitution of (4) into (3) assuming 147 zero initial conditions and neglecting the initial time  $t' < M/f_e$ , yields the solution of the time coefficients of 148 the  $r^{\text{th}}$  sub-problem according to Duhamel's integral,

$$\tau_{\rm r}^{i}(t') = \int_{0}^{M/f_e} \int_{-l/2}^{l/2} p_{\rm r}(x,\tau) \, \hat{X}_{i}^{3}(x) \hat{X}_{i}^{3}(z=h) \, \sin(\omega_{i}(t'-\tau)) \, dx \, d\tau \quad . \tag{5}$$

150 Equation (5) can be expressed as a sum of harmonic waves,

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$$\tau_r^i(t') = A\cos(\omega_i t') + B\sin(\omega_i t')$$

and after straightforward manipulations and omitting index i for simplicity in the notation, the displacement field can be expressed in the form of forward and backward travelling waves,

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$$u_r = A_s A_r e^{j(kx' \mp \omega t' - \psi)}$$

155 where only the real part of the expression is kept. The term  $A_s$  appears in the form

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$$A_{s}(\omega, \mathbf{k}) = \frac{1}{2\omega} \hat{X}^{3}(h)^{2} \sqrt{\left(\int_{0}^{M/f_{e}} h(t)\cos(\omega t) dt\right)^{2} + \left(\int_{0}^{M/f_{e}} h(t)\sin(\omega t) dt\right)^{2} \left|\int_{-w/2}^{w/2} g(x)\cos(kx) dx\right|}.$$

157 Term  $A_s$  fully describes the modal amplitude distribution from the response of a single element. It can be 158 decomposed in three terms; the first is known as the excitability function, defined as

159 
$$E = \frac{1}{2\omega} \frac{X^3(h)^2}{g(X,X)}$$

This term is related to the properties of the waveguide and cannot be altered. The normalised excitability function for the first 10 guided wave modes on a 10 mm thick aluminum plate is shown in Figure 2a. The modes exhibit a transient behavior until they reach a local maximum, after which the amplitude is strictly decaying. At high frequencies, modes A0 and S0 combine and form the Rayleigh wave [18], whose excitability remains constant with frequency. Figure 2b displays the excitability of the same modes at 2 MHz. Neglecting the Rayleigh wave, the excitability is strictly increasing from mode A1 to A5. The second term is the frequency spectrum, which can be interpreted as the norm of the Fourier transform of h(t),

167 
$$\left\|\mathcal{F}(\mathbf{h}(t))\right\| = \left\|\int_{-\infty}^{\infty} \mathbf{h}(t) \mathrm{e}^{-\mathrm{i}\omega t} \mathrm{d}t\right\| = \sqrt{\left(\int_{0}^{M/f_{e}} h(t) \cos(\omega t) \, \mathrm{d}t\right)^{2} + \left(\int_{0}^{M/f_{e}} h(t) \sin(\omega t) \, \mathrm{d}t\right)^{2}} \quad .$$
 The

168 temporal profile of the applied load h(t) can be seen as the convolution of the excitation signal and the impulse 169 response of an array element. The third term is called single element excitation spectrum, given by

170 
$$H_{se} = \left| \int_{-w/2}^{w/2} g(x) \cos(kx) \, dx \right|$$

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171 This term depends on the element width and spatial pressure distribution profile. Assuming piston-like 172 excitation, the single element spectrum for element width 1 mm is shown in Figure 3a. The maximum value is 173 reached at k = 0 rad/mm, while several local minima and maxima occur in the displayed range. However, Lamb 174 wave modes around 20 MHz·mm propagate at wavenumbers below 4 rad/mm. The effect of element width in 175 this range is shown in Figure 3b. The amplitude of high-wavenumber modes is reduced with increasing element 176 width. For example, at k = 3.5 rad/mm, the amplitude drop is -1.1, -5 and -14.5 dB for element width 0.5, 177 1 and 1.5 mm, respectively. Therefore, it is expected that more energy is distributed to lower wavenumber 178 modes when increasing element width.

179 Linearity of (3) permits the expression of the total displacement field as a superposition of the individual 180 solutions of each element. These solutions are then transformed to the global frame and summed according to

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$$u = \sum_{r=0}^{N-1} u_r = \sum_{r=0}^{N-1} A_s A_r e^{j(kx' \mp \omega t' - \psi)} = A_s \sum_{r=0}^{N-1} A_r e^{j(kx - krs \mp \omega t \pm \omega r\tau - \psi)} = N A_s H_{\pm} e^{j(kx \mp \omega t - \psi - \phi)}$$

182 where

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$$H_{\pm}(\omega,k) = \frac{1}{N} \sqrt{\sum_{r=0}^{N-1} A_r^2 + 2\sum_{r=0}^{N-1} \sum_{q=0}^{r-1} A_r A_q \cos((ks \mp \omega\tau)(r-q))}$$
(6)

and  $\psi$ ,  $\phi$  are phase values and do not affect modal amplitudes. The above form of the spectrum holds for linear time delay law and arbitrary apodisation profile. It is important to note that the modal amplitudes can be decomposed to

$$A_{\pm} = A_S H_{\pm},\tag{7}$$

188 where  $A_s = NA_s$ . Term  $A_s$  is related solely to the response of a single element; thus, it is determined by the 189 frequency spectrum, mode excitability and element width. Multiplication by *N* is merely rescaling and thus 190 insignificant. On the other hand, the excitation spectrum is strictly related to properties of the array, namely the 191 pitch, number of elements, time delay law and apodisation profile. In the following section, the properties of 192 the excitation spectrum are investigated in detail, to derive a set of conditions for single mode excitation. 193



Figure 2. Excitability function for a 10 mm thick aluminium 6082-T6 plate. a) Excitability function of the first
10 modes. b) Excitability function of the same modes at 2 MHz.

- 6 -



Figure 3. Single element excitation spectrum. a) Excitation spectrum for element width 1 mm. b) Excitation
 spectrum for element width w=0.5,1 and 1.5 mm. Higher wavenumbers are attenuated with increasing element
 width.

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## **3.** Selection of excitation parameters to enhance the purity of a single mode

If uniform amplitudes  $A_r = 1$  are considered, the excitation spectrum (6) simplifies to

$$H_{\pm}(\omega, k ; N, s, \tau) = \left| \frac{\sin\left(\frac{1}{2}N(ks \mp \omega \tau)\right)}{N\sin\left(\frac{1}{2}(ks \mp \omega \tau)\right)} \right|.$$
(8)

The spectrum depends on angular frequency and wavenumber, while the number of elements, pitch and time delay law are treated as parameters. Furthermore, using the transformations  $\omega = 2\pi f$ ,  $c_p = \frac{\omega}{k}$  and  $k = \frac{2\pi}{\lambda}$ , equivalent representations such as  $H_{\pm}(f, c_p; N, s, \tau)$  can be obtained. To enhance the forward propagation of a mode, say at  $(f_e, c_{pe})$ , the time delay constant is selected to maximise  $H_{\pm}(\omega_e, k_e)$ , requiring the denominator of the same expression to vanish [28],

$$\sin(\frac{1}{2}(k_e s - \omega_e \tau)) = 0 \Rightarrow \tau = \frac{s}{c_{pe}} - \frac{n}{f_e},$$
(9)

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is shown. The white lines represent dispersion curves of a 10 mm thick aluminium sample. b) Gated signal at
2 MHz, showing the main, grating and side lobes.

- The excitation spectrum  $H_+(f, c_p; N, s, \tau)$  for a 16-element array with pitch 4.5 mm on top of the dispersion curves (in white) of a 10 mm thick aluminium plate is shown in Figure 4a. Although the pitch value is not
- 217 representative of a conventional linear array probe, it is selected here to make it simpler to illustrate the attributes
- of the spectrum, namely the main, grating and side lobes. The time delay constant is set to  $\tau = s/c_{pe} = 1194$
- ns to enhance S3 with  $(f_e, c_{pe}) = (2 \text{ MHz}, 3.77 \text{ m/ms})$ . The excitation beam at constant phase velocity is
- shown (main lobe), while several excitation beams appear below and above  $c_{pe}$  (grating lobes). Fortunately, not
- all these beams are relevant to guided wave excitation. This is based on the fact that all guided wave modes lie between a well-defined region bounded by the Rayleigh wave  $V_R$  and cutoff velocity  $V_{\text{cutoff}}$ ,

$$V_R \leq c_p \leq V_{\text{cutoff}}$$

thereafter referred to as the excitable phase velocity region. Therefore, beams outside this region do not intersect with a guided wave mode and thus have no effect on guided wave excitation. For common steel and aluminium samples,  $V_R \approx 3$  m/ms and  $V_{\text{cutoff}} \approx 22$  m/ms. The excitation spectrum captured by the gate at  $f = f_e$  is shown in Figure 4b. The spectrum consists of the main, side and grating lobes. Note that the display dynamic range in Figure 4a thresholds all but the biggest amplitude side lobes.

The features of the excitation spectrum in Figure 4b are familiar from bulk wave steering. In fact, direct comparison of (8) with the directivity function [39] for phased array steering  $H_2(\theta)$  reveals that setting

$$sin\theta = \pm \frac{c_p \tau}{s}$$
 and  $sin\theta_s = 1$ , (10)

232 it turns out that

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$$H_2(\theta) = H_{\pm},\tag{11}$$

where  $\theta$ ,  $\theta_s$  are the azimuthal and steering angles with respect to the centre of the array, respectively. This means the material for phased array beam steering is transferable to phased array guided wave excitation. However, the performance of the array is not tested against angle  $\theta$  but phase velocity, which is not a material constant due to the multimodal and dispersive nature of guided waves. Note also that setting  $sin\theta_s = \pm \frac{c_p \tau}{s}$  and  $sin\theta = 1$  yields identical results, however, the former choice is made.

In what follows, emphasis is placed on the main attributes of the spectrum to improve guided wave excitation. More specifically, in Section 3.1, the bandwidth of the excitation spectrum is defined and an explicit form relating its width to relevant excitation parameters is provided. In Section. 3.2, a set of conditions on the pitch of the array are derived, to eliminate the grating lobes in the excitable phase velocity region. Finally, the general form of the spectrum is re-considered and an optimised apodisation profile is derived to further enhance the purity of the targeted mode in Section 3.3.

245246 3.1 Bandwidth of excitation spectrum at constant phase velocity

At high frequency-thickness products, the low dispersion modes can be roughly approximated as straight horizontal lines in the frequency-phase velocity domain. Each mode can be associated with a phase velocity bandwidth, defined as

$$\Delta c_p^{m\pm} \equiv \left| c_p^m - c_p^{m\pm 1} \right|$$

where  $c_p^m$  the phase velocity of mode m and  $c_p^{m-1}$ ,  $c_p^{m+1}$  the phase velocity of the modes closest to m such that  $c_p^{m-1} < c_p < c_p^{m+1}$ . The above definition is useful, as it quantifies how 'far' the examined mode is from its neighbours. The phase velocity bandwidth of S3 at 20 MHz·mm is shown in Figure 5.



Figure 5. Phase velocity bandwidth of mode S3 at 2 MHz on a 10 mm aluminium plate.

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256 Next, the attention is shifted to the phase velocity bandwidth  $\Delta c_p$  of the excitation beam at constant phase

velocity  $c_{pe}$  of the spectrum  $H_+$ . Based on the above, single mode excitation requires

$$\Delta c_p/2 < \Delta c_p^{m+} and \ \Delta c_p/2 < \Delta c_p^{m-}$$
(12)

where  $\Delta c_p$  is defined as the phase velocity interval containing the main lobe together with the left and right peak side lobes, see Figure 6. More specifically, using (46) from [39] and (10) the phase velocity corresponding to the local maxima of the first three side lobes, located on the right-hand side of the main lobe reads

- 262  $c_p \approx \left(1 \frac{(2m''+1)\lambda_e}{2Ns}\right)c_{pe}, \quad m'' = -2, -3, -4$  (13)
- with corresponding amplitudes at -13.5 dB, -18 dB and -21 dB, respectively, as shown again in Figure 6. Including the peak side lobes in the definition of the bandwidth ensures the maximum amplitude outside the bandwidth region is not greater than -18 dB. Note that in the above equation, the dispersion effects are neglected. This is a valid assumption for the low dispersion modal region. For example, when S3 is targeted, the error between the exact phase velocity value corresponding to the maximum of the peak side lobe and (13) is less than 0.1%. Employing (29) from [39] and (10) and setting  $m = \pm 2$ ,  $\Delta c_p$  is given by

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$$\Delta c_p \approx \left( \left( 1 + \frac{2\lambda_e}{Ns} \right) - \left( 1 - \frac{2\lambda_e}{Ns} \right) \right) c_{pe} = \frac{4\lambda_e}{Ns} c_{pe}.$$

The above equation shows that the bandwidth of the excitation spectrum increases for increasing the wavelength 270 and phase velocity of the targeted mode, but decreases when the total length of the array increases. This can be 271 achieved by increasing the number of elements or the pitch. However, as will be shown in Section 3.2, increasing 272 273 the pitch can have negative effects, as grating lobes may appear. The phase velocity bandwidth of S3 at 20 274 MHz·mm and the bandwidth of the excitation spectrum corresponding to an array with pitch 0.75 mm operating at 2 MHz for 32, 64 and 128 elements are given in Table 1. We observe that  $\Delta c_p^{m-} < \Delta c_p^{m+}$ , which suggests 275 276 that the modal density increases for lower phase velocities. The minimum number of elements to satisfy 277 condition (12) using the same array is given by

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$$\Delta c_p/2 < \Delta c_p^{S3-} \Rightarrow N > \frac{2\lambda_e}{\Delta c_p^{S3-}s} c_{pe} \Rightarrow N > 90,$$

which suggests S3 can be excited efficiently with an array length above 68 mm.

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Figure 6. Excitation spectrum  $H_+$  at 2 MHz for a 64-element, 0.75 mm pitch array. The maximum amplitude of the peak, second and third side lobes is shown. The phase velocity bandwidth is defined as the width of the main and peak side lobes.

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Table 1 Phase velocity bandwidth at 2 MHz of mode *S*3 and a 0.75 mm pitch array with 32, 64 and 128 elements.

Mode/Spectrum	Phase velocity (m/ms)	$\Delta c_p^{S3+}$ (m/ms)	$\Delta c_p^{S3-}$ (m/ms)	$\Delta c_p/2 \text{ (m/ms)}$
S3	3.77	0.32	0.21	-
32 –element array	3.77	-	-	0.59
64 –element array	3.77	-	-	0.30
128 –element array	3.77	-	-	0.15

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## 288 3.2 Elimination of grating lobes in the excitable phase velocity region

The grating lobes correspond to local maxima of the excitation spectrum and are found by setting the denominator of the spectrum  $H_+$  to zero. Combining (33) of [39] and (10), their location is determined according to

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$$c_p(m') = \left(1 - \frac{m'\lambda}{s}\right)\frac{s}{\tau},\tag{14}$$

where m' is an integer and  $\lambda = c_p/f$ . For the general linear time delay law given in (9) the main lobe occurs for m' = n, while  $m' \neq n$  corresponds to grating lobes. Therefore, there exists an infinite number of grating lobes. Nevertheless, as stated earlier, only lobes that lie in the excitable phase velocity region are relevant. To eliminate them in this region, two necessary conditions are required. The first requires the lobe at the right m' =n - 1 (see Figure 4b) of the main lobe to occur at phase velocity higher than the cutoff,

$$c_p(m'=n-1) > V_{cutoff}.$$
(15)

The second condition requires the lobe at the left m' = n + 1 of the main lobe to occur at phase velocity lower than the Rayleigh wave velocity,

$$c_p(m'=n+1) < V_R.$$
 (16)

302 At this stage, it is convenient to transform (14) in the frequency-wavenumber domain,

 $k = \frac{\tau}{s}\omega + \frac{2\pi m'}{s} = \left(\frac{2\pi}{c_{pe}} - \frac{2\pi n}{f_e s}\right)f + \frac{2\pi m'}{s}.$  (17)

This form is advantageous as it explicitly contains the frequency, which can be set to  $f = f_e$ . In the frequency-

wavenumber domain, the excitation beams appear simply as straight lines of equal and positive slope, see Figure7.

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Figure 7. Excitation spectrum  $H_+$  in the frequency-wavenumber domain for a 32-element array with pitch 2.25 mm and target mode at  $(f_e, c_{pe})$ . The excitation beams appear as straight lines. The excitation beam for m' =0 intersects the frequency spectrum (in red) on top of **S3** mode at 2 MHz. The white lines represent dispersion curves of a 10 mm thick aluminium sample.

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312 Condition (15) reads

$$\frac{\omega_e}{V_{cutoff}} > \left(\frac{2\pi}{\lambda_e} - \frac{2\pi n}{s}\right) + \frac{2\pi (n-1)}{s} \Rightarrow s < \frac{\lambda_e}{1 - \frac{V_{phase}}{V_{cutoff}}}.$$
(18)

314 Similarly, condition (16) is stated as

$$\frac{\omega_e}{V_R} < \left(\frac{2\pi}{\lambda_e} - \frac{2\pi n}{s}\right) + \frac{2\pi (n+1)}{s} \Rightarrow s < \frac{\lambda_e}{\frac{V_{\text{phase}}}{V_R} - 1}.$$
(19)

316 Conditions (18) and (19) set an upper bound for the pitch. Exploiting the fact that  $V_{\text{cutoff}} > V_{\text{phase}}$  and  $V_R <$ 317  $V_{\text{phase}}$  for any guided wave mode, both conditions can be expressed in the more compact form s <318  $\min\left(\frac{\lambda_e}{1-\epsilon_1}, \frac{\lambda_e}{\epsilon_2}\right)$ , where  $\epsilon_1$ ,  $\epsilon_2$  are positive numbers such that  $V_{\text{phase}}/V_{\text{cutoff}} = \epsilon_1$  and  $V_{\text{phase}}/V_R = 1 + \epsilon_2$ . As an 319 example, consider an array on top of a 10 mm aluminium plate operating at 2 MHz with target mode S3. Then, 320  $\epsilon_1 = 0.17, \epsilon_2 = 0.25, \lambda_e = 1.88$  mm, and simple calculations yield

$$\min\left(\frac{\lambda_e}{1-\epsilon_1},\frac{\lambda_e}{\epsilon_2}\right) = \frac{\lambda_e}{1-\epsilon_1} = 2.27 \text{ mm.}$$

322 Condition (18) proved stricter than condition (19) at the scenario above. That is indeed the case when low 323 dispersion modes close to the Rayleigh wave are targeted. Therefore, for the cases examined in this work, only 324 condition (18) is significant.

The above conditions do not eliminate grating lobes in the backwards direction. For this purpose,  $H_{-}$  is examined. The grating lobes are located at

327

329

$$c_p = -\left(1 - \frac{m'\lambda}{s}\right)\frac{s}{\tau}.$$
(20)

328 In a similar manner, (20) is first transformed to

$$k = -\frac{\tau}{s}\omega + \frac{2\pi m'}{s} = -\left(\frac{1}{\lambda_e f_e} - \frac{n}{f_e s}\right)\omega + \frac{2\pi m'}{s},\tag{21}$$

which shows that the excitation beams related to backward propagation are also straight lines with equal but negative slope. Next, it is required that the excitation beam given by m' = -n + 1 does not intersect with the frequency spectrum inside the modal region, as shown in Figure 8 for a 128-element, 0.75 mm pitch array with

time delay set to  $\tau = s/c_{pe}$ , to enhance the forward propagation of mode  $(f_e, c_{pe})$ .

- 11 -



### 334

Figure 8. Excitation spectrum  $H_{-}$  in the frequency-wavenumber domain for a 128-element array with pitch 0.75 mm and target mode at  $(f_e, c_{pe})$ . Unidirectional propagation requires that the excitation beam for m' = 1does not intersect the frequency spectrum (in red) inside the excitable region, defined by the Rayleigh wave and cutoff lines. The white lines represent dispersion curves of a 10 mm thick aluminium sample.

This is ensured if the excitation beam assumes a higher wavenumber value that the Rayleigh wave mode at  $f = f_e$ ,  $f_e$ ,

352

$$\frac{\omega_e}{V_R} < -\left(\frac{2\pi}{\lambda_e} - \frac{2\pi n}{s}\right) + \frac{2\pi(-n+1)}{s} \Rightarrow s < \frac{\lambda_e}{1 + \frac{V_{\text{phase}}}{V_P}}.$$
(22)

In contrast to condition (18) the above condition not only necessary, but also sufficient. In other words, if it is satisfied, unidirectional propagation is enforced. Furthermore, if  $V_{\text{phase}} \approx V_R$  the condition becomes  $s < \frac{\lambda_e}{2}$ , which is a well-known condition applied in phased array beam steering [39]. Note that this condition is related only to the unidirectionality of the mode and not to propagation in the forward direction.

Although conditions (18) and (22) were derived in a systematic way, some assumptions were made. Specifically, the whole derivation is solely based on the excitation spectrum, completely neglecting the effect of the single element response. In a sense, it is assumed that the single element distributes energy uniformly across the modes, while in fact, more energy is distributed to the low wavenumber modes. Furthermore, the excitation beams and frequency spectrum are simply treated as lines with zero bandwidth. For these reasons, in practice, these conditions are expected to be slightly violated.

## 353 3.3 Apodisation

Apodisation profiles can be employed to potentially improve guided wave excitation. Standard apodisation windows such as Hanning or Blackman eliminate the side lobes but increase the width of the main lobe [29]. In this section, an apodisation profile is derived based on an optimisation process. The apodisation function is compared with common apodisation profiles in Section 4.3.

Consider the general form (6) of the spectrum  $H_+$ . First, the terms that do not depend on frequency or wavenumber are omitted. The modified spectrum is given by  $\overline{H} = \sum_{r=0}^{N-1} \sum_{q=0}^{r-1} A_r A_q \cos((ks - \omega\tau)(r - q))$ , which can be expressed in the more compact form

$$\overline{H} = \underline{A}^T \widehat{H} \underline{A},$$

362 where  $\underline{A} = (A_0, ..., A_{N-1})^T$  and  $\widehat{H}$  is a  $N \times N$  matrix with elements

$$[\widehat{H}_{rq}] = \begin{cases} \cos((ks - \omega\tau)(r - q)), q < r \\ 0, \text{ otherwise} \end{cases}$$

364 where r, q = 0, ..., N - 1. Next, an objective function that needs to be maximised is defined [10,40]

- 12 -

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373

$$\Delta = \frac{\underline{A}^T \hat{H}_{e\underline{A}}}{\underline{A}^T \hat{H}_{\underline{s}\underline{A}}},\tag{23}$$

where  $\hat{H}_e = \sum_{p=0}^{P} \hat{H}_{ep}(f_{ep}, k_{ep})$  and  $\hat{H}_s = \sum_{p'=0}^{P'} \hat{H}_{sp'}(f_{sp'}, k_{sp'})$ . The points in the frequency-wavenumber 366 domain  $(f_{ep}, k_{ep})$ ,  $(f_{sp'}, k_{sp'})$  correspond to modes to enhance and suppress, respectively. The critical points 367 368 of  $\Delta$  are found by imposing the requirement

369 
$$\partial_{A_r} \Delta = 0 \Rightarrow \left(\hat{H}_e + \hat{H}_e^{\mathrm{T}} - \lambda \left(\hat{H}_s + \hat{H}_s^{\mathrm{T}}\right)\right) \underline{A} = 0 \Rightarrow R\underline{A} = \lambda \underline{A}, \tag{24}$$

where  $R = (\hat{H}_s + \hat{H}_s^T)^{-1} (\hat{H}_e + \hat{H}_e^T)$ . Solution of (24) results in finding the eigenvalues and eigenvectors of R, 370 which in the case examined in Section 4.3 was an invertible matrix. The apodisation profile is selected as the 371 372 eigenvector <u>A<sub>opt</sub></u> which corresponds to the largest eigenvalue  $\lambda_{max}$ .

#### 374 4. Experimental results

375 The experimental setup is presented in Figure 9. Two linear 2.25 MHz Vermon arrays with 128 elements and a pitch of 0.75 mm were used in pitch-catch configuration on top of a 10 mm thick 6082-T6 aluminium 376 plate. FMC data was collected along an overall line scan of 192 mm by using each element of the receiver probe 377 as independent receivers. The length of the phased array probe is 96mm so the 192mm line-scan was captured 378 379 in two steps (2x96 mm) by manually moving the probe. The separation distance was set in the first capture 380 approximately to 200 mm edge-to-edge. The transducers were aligned with a metallic frame (not shown in 381 Figure 9). The signals were generated and received with the 32Tx/32Rx/128E FI Toolbox [41], which is an 382 array driver capable of producing arbitrary waveform signals using pulse width modulation with a 3-state pulser and digitising signals with a sampling frequency of 50 MHz. To excite a single mode, a narrowband signal was 383 384 required. For that reason, in all following results, the excitation was a 14-cycle Hanning windowed toneburst 385 centred at 2 MHz, leading to 20MHz mm operating frequency-thickness product. The excitation signal was Hanning windowed to suppress any sidelobes [15]. The selected number of cycles narrows the frequency band 386 387 of the signal approximately to 0.17 MHz at -6 dB. Although this is desired to excite a single mode, an 388 immediate downside is the decrease in spatial resolution of the travelling wave, as the length of the wave-packet increases. However, the wavelengths of the guided wave modes at 20 MHz mm are in the range 1.5 - 3 mm 389 390 for the low dispersion modes, say A1 - S5. This can compensate for the high number of cycles. All 2DFFT 391 [42] results are presented on top of the dispersion curves (in white) for the 10 mm thick aluminium sample that 392 was used for the experiments.



394



## 1 m x 0.3 m x 10 mm aluminium plate

395 Figure 9. Experimental setup, showing two linear 2.25 MHz arrays with 128 elements operating in pitch-catch 396 technique on top of a 10 mm aluminium plate.

397 All signals were reconstructed synthetically [43,44]. More specifically, given the full matrix of signals 398  $[s_{ra}(t)]$ , where index r and q denote the transmitting and receiving elements, respectively, the linearity of (3) 399 suggests the reconstructed signal at  $x_a$  is given by

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$$u(x_q, h, t) = \sum_{r=0}^{N-1} A_r s_{rq} (t - t_r).$$

This means once the full matrix is obtained, any delay law and apodisation profile can be constructed synthetically. In this work, only linear time delays are considered; however, general voltage amplitudes  $A_r$  are addressed in Section 4.3.

404 The 2DFFT of a single element of the array is shown in Figure 10. Considering the amplitude 405 decomposition (7), this plot is related to term  $A_{s}$ . Essentially, the single element response sets an amplitude 406 floor, which is then enhanced or suppressed by the excitation spectrum. Simple examination reveals more energy is placed at the low wavenumber modes, which are in general more dispersive, such as A8 and S8, while little 407 408 energy is distributed to higher wavenumber modes, such as A1 and S1. The 2DFFT map reveals that there is 409 an offset between the experimental data and the dispersion curves, especially for high wavenumber modes. This error is expected to be related to the frequency-wavenumber discretisation involved in the 2DFFT [45]. The 410 411 numerical error was found around 3% and did not raise any issues, thus was deemed acceptable for the purposes 412 of this work.

In the experimental results that follow, mode S3 at  $(f_e, k_e) = (2 \text{ MHz}, 3.34 \text{ rad/mm})$  was targeted. Based on the excitability function presented in Figure 2b, S3 is more excitable that any lower order mode at 20 MHz·mm. Furthermore, this mode is less dispersive compared to higher order modes such as  $A5, S5, A6, \dots, A9$ . Modes A4 and S4 exhibit also low dispersion behavior and good excitability. However, mode S3 was preferred

417 due to its lower wavelength at 2 MHz, namely 1.89 mm, over 2.03 and 2.21 mm of A4 and S4, respectively.

These modes are also considered in Section 4.3. In all that follows, modes *S*3, *A*4 and *S*4 were excited using (9)

419 with n = 0.



#### 420



## 423

## 424 *4.1 Experimental investigation of the influence of the excitation spectrum's bandwidth to single mode* 425 *excitation*

Three setups are presented, varying the number of elements of the transmitter array, to investigate experimentally the effect of the excitation spectrum's bandwidth on guided wave excitation. In all setups, uniform amplitudes  $A_r = 1$  and time delay constant  $\tau = 200$  ns were employed. The excitation bandwidth as given in Table 1 is plotted in red on top of the 2DFFT results presented in Figure 11a, b and c. In the first configuration, only the first 32 elements of the transmitter array were used. As shown in Figure 11a, a significant amount of energy is distributed not only to S3 but also its neighbours, namely modes A3 and A4. This is expected, as the excitation bandwidth associated with a 32-element array is large enough to significantly 433 overlap with A3 and A4 around 2 MHz. In the second configuration, the first 64 elements were used for the 434 excitation. Energy leakage is reduced, and more energy is focused on mode S3, see Figure 11b. This is due to 435 the narrower spectrum bandwidth of a 64-element array. Finally, the excitation is further improved utilising all 436 128 elements and can be considered as single mode. The A-scans for the three configurations examined are 437 shown in Figure 12. The signals shown in Figure 12a, b and c were captured at a distance around 303, 307 and 438 320 mm away from the first element of the transmitter array, respectively. This way, the peak amplitude of S3 439 occurs around 122 µs for all three cases. The group velocity of A3, S3 and A4 was found to be 2.77 m/ms, 2.62 m/ms and 2.45 m/ms, respectively, with error less than 0.5%, 0.5% and 1.3% compared with the theoretical 440 441 group velocities obtained from the Dispersion Calculator [46].





**Figure 11.** 2DFFT produced with 0.75 mm pitch array and a) 32, b) 64 and c) 128 elements. The response improves significantly with the increase of number of elements, which narrows the bandwidth of the excitation





Figure 12. Example A-scans targeting mode S3 varying the number of elements a) 32 elements were used,
modes A3 and A4 are present. b) 64 elements were used, modes A3 and A4 are suppressed c) 128 elements
were used, mode S3 is dominant.

449

## 450 *4.2 Experimental verification of the generation of grating lobes*

451 Another set of experiments was conducted to examine the validity of conditions (18) and (22), which read 452 s < 2.27 mm and s < 0.84 mm, respectively.

The 2DFFT in the forward direction for s = 1.5, 2.25 and 3 mm is shown in Figure 13a, b and c. The data 453 454 was obtained from the same FMC dataset by skipping one, two and three elements of the transmitter array. The 455 active aperture was kept constant at 96 mm, and voltage amplitudes were uniform. For s = 1.5 mm, condition 456 (18) is satisfied and a single mode is generated in the forward direction, see Figure 13a. The response is very 457 similar to the one presented in Figure 11c. However, for s = 2.25 mm, an unwanted mode is generated, as 458 shown in Figure 13b. The location of the unwanted mode is at the intersection of the frequency spectrum with 459 an excitation beam (grating lobe), which can be found substituting m' = -1 in (17). This means condition (18) 460 is violated somewhat earlier in practice, which is expected, as both the frequency and the excitation beams are 461 associated with finite bandwidths. The effect of the m' = -1 beam is present also for s = 3 mm, as shown in 462 Figure 13c. The increase of the pitch has shifted the beam to the left, thus compared to the previous case, a

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higher wavenumber unwanted mode is generated.

464 Next, the 2DFFT for guided wave propagation in the backward direction is presented in Figure 13d, e and 465 f, for s = 1.5, 2.25 and 3 mm, respectively. This time, the time delay law applied synthetically is given by  $t_r = (N - 1 - r)\tau$ , to mimic a beam propagating in the -x direction. In what follows, the 0 dB value 466 corresponds to the maximum amplitude of the forward propagating wave. For example, the 0 dB value in Figure 467 13d corresponds to the maximum amplitude of Figure 13a. Note that for s = 0.75 mm condition (19) is 468 469 satisfied and no energy higher than -18 dB was found to propagate in the backwards direction, thus this result is not presented. For all three setups, condition (19) is violated and unwanted modes propagate. These are 470 471 located at the intersection between the frequency spectrum and the excitation beams given by (21). Specifically, 472 Figure 13 d, e and f show the propagating modes on top of the excitation beams for m' = 1, m' = 2 and m' = 1473 2,3, respectively. The above results are in striking agreement with the material developed in Section 3.2.



474 Figure 13. Top Row: 2DFFT in the forward direction, varying the pitch value; red line indicates an excitation 475 beam. a) s = 1.5 mm. No excitation beams intersect with the frequency spectrum and only S3 is excited. b) 476 s = 2.25 mm. The excitation beam for m' = -1 intersects with the frequency spectrum, generating low wavenumber modes. c) s = 3 mm. The same beam is activated. This time, the beam is shifted to the left, 477 478 resulting in excitation of a higher wavenumber mode. Bottom Row: 2DFFT in the backward direction, varying the pitch value. d) s = 1.5 mm. The excitation beam for m' = 1 intersects with the frequency spectrum. e) s = 1.5479 480 2.25 mm. Although the excitation beam for m' = 1 beam does not activate any modes, the beam for m' = 2does. f) s = 3 mm. Two excitation beams intersect with the frequency spectrum inside the modal region, 481 leading to the excitation of lower and higher wavenumber modes. 482

483 Similar results can be obtained if the elements of the array are grouped to simulate larger elements. For 484 example, instead of skipping two elements, the array can be partitioned into groups of three elements, combining  $1 - 3, 4 - 6, \dots, 124 - 126$ . The two approaches are equivalent in the sense that both give the same pitch value. 485 486 However, the response of a single group is different from the single element response shown in Figure 10, due 487 to the larger width of the group. The 2DFFT of a group that consists of three elements is shown in Figure 14. 488 The energy distribution to the higher wavenumber modes is below -18 dB. This makes targeting these modes 489 more challenging, mainly for three reasons. First, even if condition (18) is satisfied and no grating lobes are 490 present, a small amount of energy may still be distributed to low wavenumber modes, as the group element 491 response enhances these modes significantly. Second, the energy of the high wavenumber targeted mode is 492 decreased compared to the case where the elements are not grouped, leading to lower signal to noise ratio. 493 Comparison of the 2DFFT result shown in Figure 15a with Figure 11c revealed that the amplitude of S3 was 10 time lower. Finally, if condition (18) is violated, the unwanted modes are expected to dominate the signal.

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#### 496

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**Figure 14.** 2DFFT of a group of 3 elements. More energy is distributed to the low wavenumber modes.

499 Figure 15a, b and c present 2DFFT results in the forward direction grouping two, three and four elements, 500 respectively. All remaining parameters were kept identical to the results already presented in Section 4.2. The 501 2DFFT for pitch value 1.5 mm is shown in Figure 15a. Similar to Figure 13a, most of the energy is focused 502 on S3. However, in this case, energy leakage towards low wavenumber modes is present. Figure 15b and c 503 correspond to pitch values of 2.25 and 3 mm. Again, these are similar to Figure 13b and c, but with a sharp 504 difference. Specifically, the energy of the targeted mode S3 is significantly lower compared to energy level of 505 the unwanted modes, which dominate wave propagation, being approximately 18 dB higher than S3. This is 506 expected, as the single group response amplifies low wavenumber modes, see Figure 14. The results for 507 propagation in the -x direction are very similar to the ones already presented in Figure 13d, e and f thus are not 508 repeated here.





510

Figure 15. 2DFFT in the forward direction, varying the pitch value; red line indicates an excitation beam. a) s = 1.5 mm. Mode S3 is dominant. b) s = 2.25 mm. The excitation beam for m' = -1 intersects with the frequency spectrum, generating low wavenumber modes. Mode S3 is below -18 dB. c) s=3 mm. The same beam is activated. Mode S3 is approximately 18 dB lower.

## 515 *4.3 Experimental assessment of apodisation*

The experimental results presented so far assume uniform voltage amplitudes across the elements of the array. This raises a natural question, whether an apodisation profile can improve the purity of the targeted mode. - 17 -

518 Here, three apodisation profiles are compared, namely uniform, Blackman window and the optimised, as shown 519 in Figure 16a. The optimised profile is the result of the optimisation problem described in Subsection 3.3, where 520 the points in the frequency-wavenumber domain to enhance and suppress are manually selected based on Figure 521 11c. In general, the definition of a suitable objective function is a subtle issue; different apodisation profiles result from different definitions. Here, the points to enhance and suppress are the points of maximum amplitude 522 523 of S3 and A4, which in the frequency-phase velocity domain are given by  $(f_{e_1} c_{pe})$  and  $(f_s, c_{ps}) =$ 524 (2.08 MHz, 3.956 m/ms). The effect of each of the three profiles on the phase velocity spectrum gated at 2 525 MHz is shown in Figure 16b, similar to Figure 4b but plotted in dB scale. As expected, the Blackman window 526 significantly reduces the amplitude of the side lobes; however, the width of the main lobe increases. On the 527 contrary, the optimised profile's spectrum is very similar to the uniform spectrum, with one key difference: the second side lobe's amplitude is significantly reduced. The peak of this side lobe is at  $c_p = 3.959$  m/ms, which 528 529 is very close to  $c_{ps}$ . The amplitude of A4 at  $(f_s, c_{ps})$  depends on the magnitude of the excitation spectrum at the same point. This means that the second side lobe is primarily responsible for the generation of A4, and it is 530 531 exactly the same lobe that is suppressed by the optimised apodisation profile. Therefore, by reducing the 532 magnitude of the second side lobe of the spectrum, the amplitude A4 is expected to decrease.



Figure 16. a) Uniform, Blackman and optimised apodisation profiles for 128-element array. b) The excitation spectrum for each of these profiles is shown. The response of the optimised and uniform profiles is very similar, but the second side's lobe amplitude is significantly reduced for the optimised profile, which is beneficial to enhance the purity of *S3*.

The effect of the selected profiles on guided wave excitation is further investigated. Specifically, an
experimental 2DFFT for each amplitude profile is computed and shown in Figure 17. The overall improvement
is visible in the 2DFFT graph.





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545 The performance of each profile is further measured with the definition of an objective function which is similar 546 to (23), defined by

$$\Delta^{E} = \frac{\sum_{p=0}^{3} A^{E}(f_{ep}, k_{ep})}{\sum_{p'=0}^{14} A^{E}(f_{sp'}, k_{sp'})},$$

but in this case  $A^{E}(f_{ep}, k_{ep})$ ,  $A^{E}(f_{sp'}, k_{sp'})$  represent amplitudes extracted directly from experimental 2DFFT 548 549 results. For this calculation, four points of S3 and fifteen points of A4 were selected around 2 MHz. Then, the amplitude (linear scale) of these points was extracted from the 2DFFT for the cases of uniform, Blackman and 550 optimised apodisation profiles. The experimental objective function together with its numerator and 551 552 denominator is shown in Figure 18. The amplitude of the target mode for uniform amplitudes excitation was set to one and all others were scaled accordingly. As expected, the optimised apodisation profile performs better 553 554 among the selected three, suggesting it is the most appropriate choice for single mode excitation.





Figure 18. Amplitude comparison between Uniform, Blackman and Optimised profiles. 556

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558 Next, the possibility of further improving single mode excitation using apodisation is investigated for more 559 higher order modes, namely A4 and S4.

560 Figure 19 displays 2DFFT maps in the frequency-wavenumber domain. Uniform voltage amplitudes were 561 employed. At 2 MHz, the wavenumber values of A4 and S4 read 3.109 rad/mm and 2.814 rad/mm, thus the 562 required time delay constant is 185 ns and 167 ns, respectively. However, since the sampling period is 20 ns, 563 the applied time delays need to be rounded to the closest multiple of 20, leading to 180 ns and 160 ns. This introduces an error of 5 ns and 7 ns. To bypass this issue, the excitation frequency for mode A4 was shifted to 564 565 1.94MHz. This way, energy is centered at  $(f_{A4}, k_{A4}) = (1.94 \text{ MHz}, 2.95 \text{ rad/mm})$ , as shown Figure 19a. 566 Targeting this point requires a time delay constant  $\tau_{A4} \approx 180 ns$ . Most of the energy is focused on A4, while 567 some energy leakage is observed to modes S4 and S5. A similar frequency shift may be employed to excite S4 at 160 ns. However, to illustrate the effect of applying a time delay law slightly different from the nominal one, 568 mode S4 was excited at 2MHz. The maximum energy was found at point  $(f_{S4}, k_{S4}) = (1.93MHz, 2.58 rad/sector)$ 569 570 mm), see Figure 19b. Although the target mode is excited, in this case, mode A5 is significantly excited, 571 spanning a wide range of frequencies, from 1.95 to 2.15 MHz.

Single mode Lamb wave excitation at high frequency thickness products using a conventional linear array transducer





Figure 19. 2DFFT in frequency-phase velocity domain using uniform amplitudes and targeting modes a) A4,and b) S4.

576 The performance of uniform, Blackman and optimised profiles is tested on the selected modes. The 577 optimised profile for mode A4 is shown in Figure 20a. An objective function according to (23) was defined. The point to enhance was selected as the point of maximum amplitude of A4. The points to suppress were 578 579 selected as the maxima points of the first three side lobes of the excitation spectrum at 1.94 MHz. Figure 20b 580 shows the optimised profile for mode S4. Again, the point to enhance was selected as the point of maximum 581 amplitude of the targeted mode. The maxima of the first five side lobes at the same frequency were suppressed. 582 These maxima occur at  $c_p = 4.88, 5.01, 5.15, 5.30, 5.46$  m/ms. All these side lobes contribute to the excitation 583 of the unwanted mode A5, as it spans a phase velocity range, from 4.7 to 5.4 m/ms. In this case, the Blackman 584 window is expected to yield satisfactory results, as it effectively suppresses all side lobes. 585



Figure 20. Optimised apodisation profile for A4 and S4 modes. a) A4 apodisation profile, b) S4 apodisation
 profile.

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Figure 21a, b and c present 2DFFT results for uniform, Blackman and optimised apodisation profiles when the target mode is A4. Visual examination reveals that the optimised profile performs best, see Figure 21c, while the uniform and Blackman window have similar behavior. However, this is not true when targeting mode S4, see Figure 21d, e and f. In this case, the Blackman window yields better results, as shown in Figure 21e. This is not surprising, as the Blackman window suppresses all side lobes, and in this case, multiple sidelobes contribute to the generation of A5. The amplitude of the relevant modes extracted from the 2DFFT 595 graphs is shown in Figure 22. The procedure followed was similar to the one described for Figure 18. 596 Specifically, eight points along A4 and fifteen points along S4 and S5 around 1.94 MHz were selected for 597 Figure 22a, while seven point along S4 and seventeen points along A5 around 1.93 MHz were selected to 598 produce Figure 22b. The results agree with the previous visual observations.





Figure 21. 2DFFT results for different apodisation profiles. Top Row (A4): a) Uniform amplitudes. b)
Blackman window. c) Optimised profile. Bottom Row (S4): a) Uniform amplitudes. b) Blackman window. c)
Optimised profile.





Figure 22 Amplitude comparison between Uniform, Blackman and Optimised profiles. a) Mode A4 is targeted.
b) Mode S4 is targeted.

606

## 607 5. Conclusion

This study focused on the proper selection of key parameters for single mode excitation in the higher order mode region with a conventional linear array. An analytical model based on modal analysis was derived, showing the feasibility of the method even for guided wave related problems. An important decomposition 611 between the excitation spectrum and the response of a single element was made. Then, a set of conditions was 612 derived, see Table 2. These can be seen as guidelines for the proper selection of the number of elements, pitch and apodisation profile. The first condition determines the phase velocity bandwidth, and provided a pitch 613 614 decides the number of elements. The second and third conditions are related to eliminating grating lobes in the 615 forward and backward direction, respectively. Violation of the second condition will lead to propagation of unwanted modes, since a grating lobe will intersect with the frequency spectrum, generating low wavenumber 616 617 modes. However, satisfaction of the same requirement does not ensure single mode excitation; for example, 618 unwanted neighbouring modes might be generated, due to the small aperture length of the array. This means 619 this condition is only necessary. On the contrary, the third condition is also sufficient, since it ensures 620 unidirectional propagation. This condition is stricter that the necessary condition, in the sense that if it is satisfied, the necessary condition is satisfied as well. Therefore, it can be used to determine the pitch of the 621 array. More specifically, the pitch should be small enough to satisfy this requirement, but large enough to 622 623 increase the total length of array and thus decrease the excitation spectrum's bandwidth. Finally, the apodisation 624 profile was optimised to possibly enhance the purity of the targeted mode. The exact form of the objective 625 function is defined online, after performing 2DFFT analysis. This can be seen as a calibration process, tuning 626 each voltage amplitude to improve guided wave excitation and propagation.

627 An experimental FMC dataset was obtained, to validate the above theoretical results. This allowed different number of elements and pitch values to be evaluated after post processing from the same raw data 628 629 signals. The agreement between theoretical and experimental results was strong. Emphasis was placed on exciting S3, although modes A4 and S4 were targeted as well. The performance of different apodisation profiles 630 631 was evaluated for each of these modes. The suitability of the apodised window depended on the energy 632 distribution of the unwanted modes in the 2DFFT map. When they spanned a narrow frequency range, the 633 optimised profile gave better results compared to the uniform amplitudes and Blackman window. However, 634 when the unwanted mode spans a wider frequency range, multiple side lobes need to be suppressed. In this case, the Blackman window was found to perform best. The described setup integrated into a roller probe and its 635 636 capability to excite a pure mode in the high frequency thickness product region could be a subject to be 637 investigated in future work.

638

Condition No.	Condition	Description
1	$N > \frac{2\lambda_e}{\Delta c_p^{m-s}} c_{pe}$	Suppression of neighbouring modes of targeted mode
2	$s < rac{\lambda_e}{1 - rac{V_{phase}}{V_{cutoff}}}$	No grating lobes in $+x$ direction
3	$s < rac{\lambda_e}{1 + rac{V_{phase}}{V_R}}$	No grating lobes in $-x$ direction (unidirectional propagation)
4	$max(\Delta)$	Further enhancement of the purity of the targeted mode

630	Tabla 2	Conditions	for	single	mode	excitation
039	I able 2	Conditions	IOF	single	mode	excitation

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