

# Predictions for Combined In-Line and Cross-Flow VIV Responses with a Novel Model for Estimation of Tension

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## Abstract

The dynamic responses of slender cylinders with high aspect ratios undergoing vortex-induced vibrations (VIV) are studied. In detail, a three-dimensional model predicting the VIV responses in both the In-Line and Cross-Flow directions of slender cylinders is proposed based on the nonlinear equation governing the dynamic deformation and a wake oscillator. The tension in the cylinder is estimated according to the incoming stream velocities. To predict the VIV responses, the cylinder is discretized into finite segments, and the vibrations of each segment are estimated from solving the governing equation when the excitation forces are modelled using the Van Der Pol's wake oscillator. Considering that the wake oscillator model estimates the excitation forces according to the dynamics of the cylinder, it reveals the interactions between the flow and the dynamics of the cylinder. In order to verify the model calculating the mean tension, the VIV responses, which has been experimentally tested, is numerically studied. The comparison between the numerically predicted and experimentally measured responses shows that, the approach, especially the novel tension model, proposed herein is reliable as the frequency

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of vibrations, dominant mode number and vibration amplitude are all in good agreement with the experimental measurements and results from peer-reviewed publications.

**Key words:** Vortex-induced vibration; Van der Pol; Tension; Combined inline-crossflow responses

## 1. Introduction

Cylinders with high aspect ratios are widely used in the field of ocean engineering, especially in the offshore structures installed in deep waters Wu et al. (2012). For example, the risers widely used in the oil and gas exploration in the ocean are considered as slender cylinders in their structural designs. Along with the exploration moving from nearshore areas to deep waters, the aspect ratio of the riser connecting the floating platform and subsea production system increases significantly. In addition, the slender cylinder is also widely used to model the tendons of the tension leg platform (TLP). Given the operation water depth of the TLP (300m~1500m), the aspect ratio of the tendon is normally with an order of 10000. It is commonly acknowledged that the fatigue of the riser and the TLP tendon is the major cause for their structural failure. Considering the cost for maintaining the offshore structures and the subsea systems, the structural failure of the riser or the tendon of the TLP could lead to huge economic losses. Consequently, the vibrations of cylinders with high aspect ratios are of primary concern for their structural designers.

Instability of flow around cylinders induces regular vortex formation and asymmetric vortex shedding, which produces periodic fluctuations in lift forces acting on the cylinders (Bearman et al., 2001). Given fluctuating lift forces, the cylinder vibrates at the frequencies determined by the interaction between the flow and the dynamics of the cylinder. When the frequency  $\Omega_f$  of vortex shedding, which determines, to a large extent, the vibration frequency of the cylinder, approaches the natural frequency  $\Omega_s$  of the cylinder, vibration amplitude of the cylinder is enlarged and  $\Omega_f$  is locked onto  $\Omega_s$  for a range of stream velocities (Stansby, 1976). Due to the large resonance vibration amplitudes and the lock-in effect, the vibrations induced by the vortex shedding, which is referred to as Vortex-Induced Vibrations (VIV) hereafter, could bring serious

42 damages to the cylinder, and hence the offshore structure in which the slender cylinder is an essential piece. Even if the  
43 resonance is avoided, the long-term VIV leads to the fatigue of the riser, which ultimately damages the system. Therefore,  
44 VIV of slender cylinders has long been a focus for the academic investigations.

45 Vortex shedding is a three-dimensional (3D) process, which makes cylinders vibrate in both in-line (IL) and cross-flow (CF)  
46 directions. The VIV of slender cylinders could be investigated through simulating the dynamics of elastically mounted short  
47 rigid cylinders (low aspect ratio) in experiments. In such experiments, only the vibration in the cross-flow direction is  
48 commonly investigated (Dahl et al., 2006). Recent studies, however, have found that long flexible cylinders acquire different  
49 dynamic responses from those found in the vibrations of short rigid cylinders. Firstly, long flexible cylinders usually vibrate  
50 at higher frequencies and in higher modes comparing to the short rigid ones (Trim et al., 2005). Secondly, both standing wave  
51 pattern and traveling wave pattern are observed for long flexible cylinders while only standing waves relate to the vibrations  
52 of short rigid cylinders. Because Vandiver et al. (2009) found that most of the VIV energy was concentrated in traveling waves,  
53 the difference in vibration patterns could lead to unrealistic estimates of VIV responses for the long flexible cylinder based  
54 on experimental results of short rigid cylinders. Thirdly, in the lock-in region, short rigid cylinders vibrate in only one mode  
55 because modal frequencies are well separated for the short rigid cylinders (Iwan and Jones, 1987), but vibrations with  
56 combined modes are frequently observed for long flexible cylinders. Fourthly, when the excitation frequency ratio (IL  
57 vibration frequency  $f_{e,IL}$  to CF vibration frequency  $f_{e,CF}$ ) equals the natural frequency ratio (IL natural frequency  $\Omega_{s,IL}$  to CF  
58 natural frequency  $\Omega_{s,CF}$ ), dual resonance is observed for the long flexible cylinders but not properly simulated in the  
59 experiments with short rigid cylinders (Dahl et al., 2006; Dahl et al., 2010). More importantly, some studies suggested that IL  
60 vibrations are as important as CF vibrations for long flexible structures. For example, Sarpkaya has found that the combination  
61 of IL and CF vibrations in a two degree-of-freedom (i.e. 2DOF) system could produce larger amplitude VIV responses  
62 (Sarpkaya, 1995) compared to the VIV considering only CF direction dynamics. Such findings are in agreement with the  
63 conclusion drawn by Moe and Wu (1990) and others that large CF amplitude occurred in a wider range of reduced velocities

64 due to energy transforming from IL motions to CF motions (Dahl et al., 2006). Trim et al. (2005) conducted a benchmark  
65 experiment investigating the IL and CF VIV responses under uniform and shear flow conditions and found comparable  
66 magnitude of fatigue damage in both CF and IL directions. Blevins and Coughran have found that a 2DOF model of the long  
67 flexible cylinders has larger velocity entrainment (sometimes referred to as the synchronization, lock-in or lock-on) band than  
68 the model containing only CF vibrations (Blevins and Coughran, 2009). That means 2DOF response is larger at constant  
69 reduced damping and mass ratio. Therefore, further investigations on the dynamics of the long flexible cylinders, with  
70 emphasis on the combined CF and IL vibrations, are necessary for predicting its VIV responses.

71 There are generally two approaches to numerically predict VIV of slender cylinders, i.e. computational fluid dynamics (CFD)  
72 techniques and semi-empirical methods in which forces exerted on the oscillating cylinders are estimated via a semi-empirical  
73 model. While the CFD simulation explicitly produces all the details of the flow around the cylinder, which in turn yields  
74 reliable estimations of drag and lift forces exerted on the cylinder, the semi-empirical model estimates drag and lift forces  
75 according to the data measured in experiments (Wu et al., 2012). Although the CFD simulation produces more accurate and  
76 reliable estimations of drag and lift forces, it requires much more computational resources for the case with realistic Reynolds  
77 numbers when comparing to the semi-empirical method. In addition, the CFD simulation results are not realistic for predicting  
78 VIV for long flexible structures (Xu et al., 2008). Semi-empirical methods, on the other hand, include various wake oscillator  
79 models and predicting VIV using the data measured in forced vibration experiments. In general, a wake oscillator model  
80 contains a dynamical system to simulate vortex shedding. It therefore helps enhance the understanding of the physics of VIV  
81 as the model provides some insights into the physical mechanism governing the flow and vortex shedding. The commercial  
82 codes developed based on the semi-empirical methods (such as VIVA, VIVANA and SHEAR7), usually build hydrodynamic  
83 coefficient database using the measurements gathered in a series of experiments and predict the VIV using the coefficients  
84 kept in the database. It should be noted that the common semi-empirical model only consider the VIV at a limited number of  
85 discrete frequencies in the CF direction. Consequently, there is still room for the semi-empirical model to be improved in

86 terms of predicting the VIV response in both the IL and CF directions of slender cylinders.

87 In the light of estimating the vibrations of cylinders under the excitations of the vortex shedding, Bishop and Hassan have  
88 suggested that the wake of the cylinder behaves as a conventional mechanical oscillator (Bishop and Hassan, 1964). Following  
89 their suggestion, Hartlen and Currie proposed a wake oscillator model, in which the fluctuating lift coefficient satisfies a Van  
90 Der Pol type equation (Hartlen and Currie, 1970). Based on the Hartlen-Currie model, Skop and Griffin devised a modified  
91 Van Der Pol equation and developed relations between empirical constants and physically meaningful parameters (Skop and  
92 Griffin, 1973). In addition, the work of Skop and Griffin (1973) contains a verification showing that the proposed model  
93 predictions are in quantitative agreement with experimental observations. Nayfeh et al. (2003) combined CFD simulation and  
94 wake oscillator assuming that drag is the function of lift as their first step in the development of a reduced order model. They  
95 found that Van Der Pol equation is suitable to model the lift compared to Rayleigh equation. Facchinetti et al. (2004)  
96 investigated three different schemes coupling motions of cylinder segments and the wake oscillator (displacement, velocity  
97 and acceleration coupling). It turned out that acceleration coupling yielded the best agreement with experimental  
98 measurements. Using the acceleration coupling scheme, Xu et al. (2008) proposed a model for high aspect ratio riser with  
99 nonlinear coupling between axial and the CF motions and compared the results with CFD results and experimental data.  
100 Violette et al. (2007), in addition, numerically solved the Partial Differential Equations (PDEs) governing the vibrations of  
101 the cylinder coupled with the Van Der Pol oscillator and compared the solution with Direct Numerical Simulation (DNS)  
102 results and experiment data. Based on the work of Nayfeh et al. (2003), Akhtar et al. (2009) developed the reduced order  
103 model (Van der Pol-Duffing model) for flow over elliptic cylinders with different eccentricities. They performed the CFD  
104 simulations first, then the CFD results was used to identify the coefficients in the reduced order model. Their model results  
105 agree well with the CFD data. Later, Srinil and Zanganeh (2012a) modelled CF and IL vibrations using double structural  
106 duffing equations-Van der Pol wake oscillators and introduced cubic and quadratic nonlinear terms to structural equations.  
107 Gu et al. (2012) applied the Generalized Integral Transform Technique to predict VIV of the slender cylinder via transforming

108 PDEs to Ordinary Differential Equations (ODEs). Stabile et al. (2018) proposed a novel reduced order model, which consists  
109 a forced Van der Pol oscillator and a linear state-space model to model the CF and IL forces. Their model matches the  
110 experimental results well.

111 It is widely acknowledged that the tensions in the risers, or other structure members with high aspect ratio, influence their  
112 VIV responses. In fact, Srinil (2011) has shown that a realistic model estimating the tensions in the slender cylinder is critical  
113 for predicting its VIV behaviors. Generally, there are three branches in terms of modelling the tension associated with a  
114 slender cylinder. The most common approach is to model the tension as a constant determined purely based on the forces  
115 applied at the end of the riser (Mathelin and de Langre, 2005; Sanaati and Kato, 2012). Such an approach is useful in the case  
116 where the riser is pre-tensioned, but the variations induced by the deformation of the riser is neglected. In addition, the tension  
117 can be modelled as a function of the height (depth), as in the studies of Srinil and Chen, Li et al. (Chen et al., 2012; Srinil,  
118 2011) to account for the losses in tension due to the buoyance. Like the constant tension model, the models employing a  
119 vertically varying tension is infeasible to account for the axial deformation found in risers due to the VIV. To model the  
120 influence of tensions in risers in a more realistic way, Ge et al. (2009) and Gu et al. (2012) introduced a model calculating the  
121 tension according to the cylinder prolongation. More specifically, the axial deformation of the riser is calculated according to  
122 the motions of the riser segments. Given that tensions are calculated according to the length of the riser, the dynamic responses  
123 of long risers can be estimated in a more realistic manner, which showed different patterns from the vibrations of risers with  
124 constant or vertically varying tensions.

125 Although the study of Gu et al. (2012) included a tension model employing the prolongation of the riser as the independent  
126 variable, it focused on the CF vibrations only. In addition, the work of Ge et al. (2009) didn't evaluate the relation between  
127 the tension and flow velocity. Thus, more efforts are needed in terms of contributing a 3D model predicting for both the IL  
128 and CF vibrations with dynamically varying tensions.

129 In the present study, a 3D VIV model is proposed, which includes a new formula calculating the tension of a long flexible

130 cylinder and considering combined IL and CF VIV responses. The formula is validated against the data presented in Trim et  
 131 al.'s work (Trim et al., 2005) and the model results are compared to the numerical simulations performed by Ge et al. (2009)  
 132 and Gu et al. (2012). After the introduction, Section 2 presents the proposed 3D model and the formula calculating the tension.  
 133 Section 3 shows a case study in which the VIV in both IL and CF directions are predicted using the proposed model. Moreover,  
 134 the results are compared to the experimental data to verify the proposed model in Section 3. The advantages and disadvantages  
 135 of the proposed model are also discussed in Section 3 based on the similarities and differences between the numerical  
 136 predictions and the experimental measurements. Conclusions are drawn in Section 4.

## 137 **2. Model description**

### 138 **2.1 Nonlinear coupled structure and wake oscillator model**

139 The physical system considered herein is a flexible beam with simple supports, modelling a riser with diameter  $D$  subjected  
 140 to a uniform flow with the stream velocity of  $U$ . The deflection of the beam is expressed by the Euler-Bernoulli beam equation.  
 141 The riser is free to oscillate both in IL direction (x-axis) and CF direction (y-axis) as shown in Fig.1. The motion governing  
 142 equation for x and y displacements at any time  $t > 0$  and position  $0 < z < L$  along the riser's length is expressed as

$$143 \quad M \frac{\partial^2 r}{\partial t^2} + R \frac{\partial r}{\partial t} + \frac{\partial^2}{\partial z^2} \left( EI \frac{\partial^2 r}{\partial z^2} \right) - \frac{\partial}{\partial z} \left( T \frac{\partial r}{\partial z} \right) = F, \quad (1)$$

144 where

$$145 \quad r = x + iy, \quad F = F_x + iF_y, \quad M = m_s + \frac{\pi}{4} C_a \rho D^2, \quad R = R_s + R_f = 2\xi_d M \Omega_s + \gamma \rho D^2 \Omega_f, \quad \Omega_s = \sqrt{\frac{\Omega_n^2}{1 + \frac{C_a}{m_s / (\frac{\pi}{4} \rho D^2)}}}. \quad (2)$$

146 The meaning of the symbols appeared here and in the following equations can be found in APPENDIX: LIST of SYMBOLS  
 147 In Equations (1) and (2), displacement  $r$  and excitation force  $F$  are represented by complexes and  $i$  is the imaginary unit.  $x$   
 148 and  $y$  are the displacements in the IL and CF directions, respectively. And excitation force  $F$  is the fluid force exerted on the

149 riser, decomposing into  $F_x$  and  $F_y$ . The external force in the  $x$  direction  $F_x$  is the sum of the mean  $f_D$  and fluctuating drag force  
 150  $f_D'$  as shown in

$$151 \quad F_x = f_D + f_D'. \quad (3)$$

152 In Equation (1), the total mass  $M$  includes structural mass  $m_s$  and added mass.  $\rho$  is the fluid density and  $D$  is the diameter of  
 153 the riser. According to Song et al., the added mass coefficient  $C_a$  varies along the riser's length significantly in both IL and  
 154 CF directions (Song et al., 2016). In the present study,  $C_a$  is taken as a constant and its variation is neglected. Damping  $R$  is  
 155 decomposed into structural damping  $R_s$  and added damping  $R_f$  as well.  $\xi_d$  is the damping ratio,  $\gamma$  is the empirical parameter  
 156 and  $\Omega_f$  is the Strouhal frequency. The representative natural frequency  $\Omega_s$  will be the natural frequency in water due to low  
 157 mass ratio of the riser (Bearman et al., 2001).  $\Omega_n$  is the natural frequency in air of the riser.  $E$  is the Young's modulus. The  
 158 rigidity  $EI$ , on the other hand, is assumed as constant along the length of the cylinder.

159 For the original structural motion equation (equation (1)), tension has been already included as a dedicated term  $(\frac{\partial}{\partial z}(T \frac{\partial y}{\partial z}))$ .  
 160 Considering the tension essentially influences the stiffness of the riser, the tension calculated according to the term in the  
 161 equation would modify the dynamics of the riser, and eventually the resulting deformation of the riser. Consequently, directly  
 162 involving the tension term as in the original equation leads to an iterative process in estimating the VIV at every time step.  
 163 Understandably, iterations within a single time step is at high computational cost for a semi-empirical model to estimate the  
 164 VIV. In order to make the simulation more efficient, tension is assumed to be a constant  $T \frac{\partial^2 y}{\partial z^2}$  in various studies (Furnes and  
 165 Sørensen, 2007; Gao et al., 2018; Gao et al., 2019; Ge et al., 2009; Ge et al., 2011). However, Gu et al. (2013) found that the  
 166 prolongation, and also tension, increases with the increasing velocity, which agrees with the experimental results. In addition,  
 167 Lee and Allen (2010) found that the increase of the vibration frequency with flow speeds is strongly related to the rise of the  
 168 axial tension. Based on their studies, tension should not be simply modeled as a constant. In the present study, a new model  
 169 calculating the tension according to the flow velocity is proposed, which will be elaborated in subsection 2.2.

170 In estimating the external forces, Vandiver modified the empirical relation for predicting the total drag coefficient  $\bar{C}_D$  as

171  $\bar{C}_D = C_{D0} \left[ 1 + 1.043 (2Y_{RMS/D})^{0.65} \right]$ , where  $C_{D0}$  is the mean drag coefficient of a stationary rigid cylinder and  $Y_{RMS/D}$  is the  
172 root mean square of the antinode displacement (Vandiver, 1983). Following the suggestion of Rosetti and Nishimoto et al.,  
173  $\bar{C}_D$  is taken as  $C_{D0} (1 + Kq^2)$  (Rosetti et al., 2009) where  $K$  is amplification factor and  $q$  is the cross flow variable which  
174 will be discussed later. Thus, mean drag force is calculated as

$$175 \quad f_D = \frac{1}{2} \bar{C}_D \rho D U^2 = \frac{1}{2} C_{D0} (1 + Kq^2) \rho D U^2. \quad (4)$$

176 The variables  $p$  and  $q$ , as shown in

$$177 \quad f'_D = \frac{1}{2} C_{Di} \rho D U^2, \quad (5)$$

$$178 \quad F_y = \frac{1}{2} C_L \rho D U^2, \quad (6)$$

$$179 \quad C_{Di} = C_{Di0} \frac{p}{2}, \quad C_L = C_{L0} \frac{q}{2}, \quad (7)$$

180 are introduced to model the fluctuating nature of drag and lift forces.

181 Here,  $C_{Di}$  and  $C_L$  are the vortex shedding drag and lift coefficients,  $C_{Di0}$  is the amplitude of vortex shedding drag coefficient  
182 and  $C_{L0}$  is the lift coefficient of a stationary rigid cylinder. The fluctuating drag and lift forces are periodic and self-limited,  
183 satisfying Van Der Pol nonlinear oscillator equations, which are expressed as

$$184 \quad \frac{\partial^2 p}{\partial t^2} + 2\varepsilon_x \Omega_f (p^2 - 1) \frac{\partial p}{\partial t} + 4\Omega_f^2 p = \frac{A_x}{D} \frac{\partial^2 x}{\partial t^2}, \quad (8)$$

$$185 \quad \frac{\partial^2 q}{\partial t^2} + \varepsilon_y \Omega_f (q^2 - 1) \frac{\partial q}{\partial t} + \Omega_f^2 q = \frac{A_y}{D} \frac{\partial^2 y}{\partial t^2}, \quad (9)$$

$$186 \quad \text{where } \Omega_f = 2\pi S_t \frac{U}{D}. \quad (10)$$

187 Here,  $\varepsilon_x$ ,  $\varepsilon_y$ ,  $A_x$  and  $A_y$  are the empirical parameters which will be stuned by the experiments The acceleration coupling  
188 scheme is adopted in the present study to link structural motion equations and Van Der Pol oscillator as suggested by  
189 Facchinetti et al. (2004).

190 A number of studies (Kurushina et al., 2018; Ogink and Metrikine, 2010; Postnikov et al., 2016) indicated that hydrodynamic  
 191 forces rely on the relative velocity between cylinder motion and free stream velocity and some studies (Ge et al., 2009; Ge et  
 192 al., 2011; Srinil and Zanganeh, 2012b; Wang et al., 2003) considered force decomposition. In this study, force decomposition  
 193 technique is adopted and the excitation forces are expressed as

$$\begin{aligned}
 194 \quad F_x &= f_D + f_D' - f_L \dot{y} \quad U \quad \frac{1}{2} C_{D\dot{y}} \dot{y} \quad Kq \dot{y} \quad DU \quad \frac{1}{2} C_{Di\dot{y}} \frac{P}{2} \quad DU \quad \frac{1}{2} C_{L\dot{y}} \frac{q}{2} \quad DU \dot{y} \\
 F_y &= f_L + f_D' \dot{y} \quad U \quad \frac{1}{2} C_{L\dot{y}} \frac{q}{2} \quad DU \quad \frac{1}{2} C_{Di\dot{y}} \frac{P}{2} \quad DU \dot{y}
 \end{aligned}
 \tag{11}$$

195 where the dot represents the derivative of the time, i.e.  $\dot{y}$  is the velocity in the y direction.

## 196 2.2 Determination for mean tension

197 Lee and Allen have conducted a series of experiments and have found that for tension-dominated riser, top tension increases  
 198 with the increasing stream velocity (Lee and Allen, 2010). In the subsection, mean tension force is derived as a function of  
 199 the stream velocity.

200 Mean tension force can be computed according to the Hooke's law showing

$$201 \quad T = T_{ini} + EA \frac{\Delta L}{L}, \tag{12}$$

$$202 \quad \text{where } \Delta L = S - L. \tag{13}$$

203 In Equations (12-13), the mean tension  $T$  is modelled proportional to the prolongation of the cylinder.  $T_{ini}$  is the initial tension,  
 204  $A$  is the cross-sectional area,  $L$  is the initial length of the riser,  $S$  is the instantaneous length and  $\Delta L$  is the prolongation of the  
 205 riser. In the literature, the instantaneous length of the cylinder  $S$  is calculated as

$$206 \quad S = \int_0^L \sqrt{1 + \left(\frac{\partial y}{\partial z}\right)^2 + \left(\frac{\partial x}{\partial z}\right)^2} dz. \tag{14}$$

207 For a long flexible cylinder, the prolongation is resulted from the deflections forced by the drag and lift forces experienced

208 by the cylinder. Considering that the force is consist of the mean drag, fluctuating drag and lift, the deflection contains  
 209 corresponding three parts. Among them, the deflection  $x_{mean}$  induced by mean drag force is typically greater than the deflection  
 210 resulted from fluctuating drag and lift forces. More importantly, in a long term simulation of VIV, the prolongation induced  
 211 by deflections corresponding to fluctuating drag and lift does not have lasting influence. More specifically, the deflections  
 212 corresponding to fluctuating drag and lift vary temporally and therefore has only impacts on limited temporal and spatial  
 213 scales. So,  $\left(\frac{\partial y}{\partial z}\right)^2$  in Equation (14) is neglected. In the case that  $x_{mean}$  is sufficiently small and its derivative is assumed to

214 be small, the instantaneous length  $S$  could be simplified as

$$215 \quad S = \int_0^L \sqrt{1 + \left(\frac{\partial x_{mean}}{\partial z}\right)^2} dz = \int_0^L \sqrt{1 + x'_{mean}} dz \approx \int_0^L 1 + \frac{1}{2} x'^2_{mean} dz, \quad (15)$$

216 where  $x'_{mean}$  represents the derivative of  $x_{mean}$  with respect to  $z$ .

217 Simplifying Equations (13-15) yields

$$218 \quad \Delta L \approx \frac{1}{2} \int_0^L x'^2_{mean} dz. \quad (16)$$

219 For a pin-ended beam, an analytical solution to Equation (16) exists if the deflection shape is expressed as a sine function (Gu  
 220 et al., 2012) showing,

$$221 \quad x_{mean} = \frac{P}{EI \frac{\pi^4}{L^4} + T \frac{\pi^2}{L^2}} \sin\left(\pi \frac{z}{L}\right), \quad (17)$$

222 where  $P$  is the external force per unit length approximated as,

$$223 \quad P = \frac{1}{2} \rho D U^2 C_D. \quad (18)$$

224 After some manipulations of Equations (12-13,16-18), the mean tension is obtained,

$$225 \quad T = T_{ini} + \frac{EA}{16\pi^2} \left( \frac{\rho D U^2 C_D L^3}{EI \pi^2 + T L^2} \right)^2. \quad (19)$$

226 Based on the tension calculated according to Equation (19), the influence of tensions on the vibrations in both the CF and IL

227 directions are investigated in the present study. Then the formula, i.e. Equation (19), is verified.

## 228 2.3 Boundary and initial conditions

229 Since the long flexible cylinder is assumed to be pinned at the ends, the displacements and moments at the ends should hence  
230 be kept zero during vibrations:

$$231 \quad r(0,t) = r(L,t) = 0 \quad (t > 0),$$

$$232 \quad \frac{\partial^2 r}{\partial z^2}(0,t) = \frac{\partial^2 r}{\partial z^2}(L,t) = 0 \quad (t > 0).$$

233 The initial conditions in solving for the displacements of the cylinder and the corresponding wake variables are, given as,

$$234 \quad r(z,0) = 0, \quad \frac{\partial r}{\partial t}(z,0) = 0, \quad (0 < z < L);$$

$$235 \quad p(z,0) = q(z,0) = 2, \quad \frac{\partial p}{\partial t}(z,0) = \frac{\partial q}{\partial t}(z,0) = 0, \quad (0 < z < L).$$

236 Given the boundary and initial conditions, the time history of displacements along the cylinder are then simulated through  
237 numerically solving the governing equations of cylinder dynamics. More specifically, the standard central finite difference  
238 scheme is employed to discretize the equations in both space and time domains. The cylinder is separated into  $N$  segments by  
239  $N-1$  points and by the distance  $h$ . Let the displacement  $r$ , at point  $i$  ( $3 \leq i \leq N-2$ ) be denoted as  $r_i$ . The superscript refers to the  
240 time step. The approximations of Equations (1, 8-9) are

$$241 \quad M \left( \frac{r_i^{(t+1)} - 2r_i^{(t)} + r_i^{(t-1)}}{(\Delta t)^2} \right) + (R_f + R_s) \left( \frac{r_i^{(t+1)} - r_i^{(t-1)}}{2\Delta t} \right) + EI \left( \frac{r_{i+2}^{(t)} - 4r_{i+1}^{(t)} + 6r_i^{(t)} - 4r_{i-1}^{(t)} + r_{i-2}^{(t)}}{h^4} \right) - T \left( \frac{r_{i+1}^{(t)} - 2r_i^{(t)} + r_{i-1}^{(t)}}{h^2} \right) \\ = F_i^{(t)}, \quad (20)$$

$$242 \quad \frac{p_i^{(t+1)} - 2p_i^{(t)} + p_i^{(t-1)}}{(\Delta t)^2} + 2\varepsilon_x \Omega_f (p_i^{2(t)} - 1) \frac{(p_i^{(t+1)} - p_i^{(t-1)})}{2\Delta t} + 4\Omega_f^2 p_i^{(t)} = \frac{A_x}{D} \left( \frac{x_i^{(t+1)} - 2x_i^{(t)} + x_i^{(t-1)}}{(\Delta t)^2} \right), \quad (21)$$

$$243 \quad \frac{q_i^{(t+1)} - 2q_i^{(t)} + q_i^{(t-1)}}{(\Delta t)^2} + \varepsilon_y \Omega_f (q_i^{2(t)} - 1) \frac{(q_i^{(t+1)} - q_i^{(t-1)})}{2\Delta t} + \Omega_f^2 q_i^{(t)} = \frac{A_y}{D} \left( \frac{y_i^{(t+1)} - 2y_i^{(t)} + y_i^{(t-1)}}{(\Delta t)^2} \right), \quad (22)$$

244 as done by Ge et al. (2009) and Gosse and Barsdale (1969).

245 With the solution for the above equations, the displacements, as a function of the z coordinate and time, are obtained showing  
246 the vibrations of the cylinder.

### 247 3. Case study

248 In this section, the experimental test case conducted by Trim et al. (2005) is simulated using the proposed numerical model  
249 to validate that Equation (19) is reliable to model the mean top tension for a long flexible cylinder. Main parameters defining  
250 the cylinder are listed in Table 1. The free-stream velocity in the simulation varies from 0.3m/s to 2.4m/s with an increment  
251 of 0.1m/s.

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Table 1 Riser characteristics (Trim et al., 2005)

Outer diameter	$D=0.027\text{m}$	Young's modulus of elasticity	$E=36.2\times 10^9\text{N/m}^2$
Inner diameter	$d=0.021\text{m}$	Axial tension	$T_{int}=4\text{-}6\text{kN}$
Wall thickness	$t_w=0.003\text{m}$	Aspect ratio	1407
Length	$L=38\text{m}$	Mass ratio	1.6
Structure mass	$m_s=0.939\text{kg/m}$	Density	$\rho=1025\text{kg/m}^3$

256 In the present study, the environmental parameters are given (See Table 2) following the suggestions in the literature (Furnes  
257 and Sørensen, 2007; Ge et al., 2009; Rosetti et al., 2009).

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Table 2 Environmental parameters

$Ca$	1	$St$	0.17
$C_{D0}$	1.2	$C_{Di0}$	0.1
$C_{L0}$	0.3	$\gamma$	0.45
$\varepsilon_x$	0.3	$\varepsilon_y$	0.3
$A_x$	12	$A_y$	12

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Figure 2 shows the variation of mean tension with stream velocity in the present simulation and in the simulation reported by Gu et al. (2012). In addition, the influence of mean tensions on the natural frequency of the cylinder, or the riser, is also presented in Figure 2. It is evident from the figure that the mean tension increases from 5kN at  $U=0.3\text{m/s}$  to 15.9kN at  $U=2.4\text{m/s}$  as predicted by Equation (19). In the simulation reported by Gu et al. (2012), on the other hand, the mean tension increases from 6kN to 10.6kN. Moreover, Figure 2 implies a linear relationship between the natural frequency and the square root of the mean tension as in experimental studies (Lee and Allen, 2010). In fact, Figure 2 substantiates that the natural frequency of the cylinder is not an inherent property of tensioned structural component but a variable reflecting the influence of external loads. The fact that the numerical simulation reported in the present study support a linear relationship between the natural frequency and the square root of the mean tension substantiates that the proposed numerical model is applicable to simulate the overall dynamics of the cylinder under the influence of tensions and environmental flows.

Besides the illustrative comparisons shown in Figures 2, Figure 3 and Figure 4 present the comparison of dominant mode numbers and vibration frequencies obtained in the present simulation and extracted from the manuscripts of Ge et al. (2009), Gu et al. (2012) and Trim et al. (2005). More specifically, Figure 3 shows that the dominant mode number increases with the stream velocity, and the number corresponding to the IL vibration is almost twice as large as the number corresponding to the CF vibration. Comparisons presented in Figure 3 indicate that the CF dominant mode numbers obtained from the present study agree with the experimental measurements and numerical results reported by Ge et al. (2009) and Gu et al. (2012). The

276 maximum deviation of mode number, from the prediction of the proposed model to the experimental results, in the IL direction  
277 is 5, smaller than the value reported by Ge et al. (2009). The maximum deviation in the CF direction, in addition, is 2  
278 comparing to 3 as reported by Ge et al. (2009) and 4 as reported by and Gu et al. (2012). In the IL direction, Figure 3 shows  
279 that the proposed model is able to capture the vibration with high mode numbers in the high-speed streams. The predictions  
280 given by Ge et al. (2009), on the other hand, considerably deviate from the experimental data. It should be noted that the IL  
281 vibrations are not included in the simulation conducted by Gu et al. (2012), and therefore no data is available for the  
282 comparison. When the free-stream velocity is within the range of 1.7-2.2 m/s, the mode numbers predicted by both the  
283 proposed model and the model suggested by Ge et al. (2009) are lower than the experiment results reported by Trim et al.  
284 (2005). In the CF direction, both the proposed model and the model suggested by Gu et al. (2012) predict vibrations with a  
285 higher mode number, in contrast to the results reported by Ge et al. (2009), when the stream velocity is in the range of 2 to  
286 2.4m/s. When the flow velocity is 1.5-2m/s, the proposed model predicts a larger value of the mean tension than the model  
287 suggested by Gu et al. (2012), resulting in a higher mode number. Consequently, the dominant mode number predicted by the  
288 proposed model is in better agreement with the experimental data given by Trim et al. (2005). When the stream velocity is in  
289 the range of 1.5-2 m/s, the dominate mode number predicted by the proposed model is, however, lower than the Trim's  
290 experimental (Trim et al., 2005) results in the IL direction, and higher than Trim's results (Trim et al., 2005) in the CF direction.  
291 Such a finding implies that the proposed model still needs improvements.

292 Figure 4 shows that frequencies of cylinder vibrations increase with stream velocities. In addition, Figure 4 reveals that the  
293 frequencies corresponding to the IL vibration is twice as large as the frequencies of the vibrations in the CF direction. For the  
294 sake of being illustrative, the vortex shedding frequency calculated according to the Strouhal relation and the doubled  
295 shedding frequency are plotted in Figure 4. It is well reported in the previous investigations that the measured/simulated  
296 frequencies of the cylinder vibration are different from the vortex shedding frequency of the fixed cylinder (Trim et al., 2005).  
297 Such differences are also observed in Figure 4. In fact, Figure 4 indicated that the vibration frequencies predicted by the

298 proposed model are in better agreement with the experimental data when comparing the numerical results given by Ge et al.  
 299 (2009). The root mean square (RMS) of deviation is 2.10 in the IL direction compared to 2.45 reported by Ge et al. (2009)  
 300 while it is 1.13 in the CF direction compared to 1.43 and 1.15 reported by Ge et al. (2009) and Gu et al. (2012), respectively.  
 301 As the dynamic responses of the cylinder are in associations with the stiffness of the structure, which is in turn influenced by  
 302 the tension, the agreement presented in Figure 5 validates the proposed model in modelling the mean tension and its influence  
 303 on the dynamics of the cylinder (Equation 19).

304 As introduced in Section 2.3, both IL and CF displacements are the function of space and time. In the verification, the  
 305 maximum of displacement standard deviation  $\xi$ , which is defined as in Equation (23), is used to quantitatively assess the  
 306 deviation of the prediction from the proposed model to the experimental data.

$$307 \quad \xi = \max(S_i), \quad S_i = \frac{std(r_i(t))}{D} = \frac{1}{D} \sqrt{\frac{\sum (r_i^{(t)} - \bar{r}_i)^2}{Nt-1}}, \quad 1 \leq i \leq N, \quad (23)$$

308 where  $N$  and  $Nt$  are total numbers of nodes and time steps, respectively,  $S_i$  represents displacement standard deviation at node  
 309  $i$  and  $\bar{r}_i$  is mean value of displacements at node  $i$ . It should be noted that data is taken after the simulation is stable, i.e.  
 310 periodic vibration with almost constant amplitude. Figure 5 shows the maximum of displacement standard deviation  $\xi$  in  
 311 CF and IL directions varying with the stream velocity. It is shown in the figure that  $\xi$  for CF and IL directions are around  
 312 0.9D and 0.2D, respectively. In addition, Figure 6 implies that the amplitude predicted by the proposed model for CF  
 313 vibrations is larger than the experimental data, which makes the predictions conservative in the assessment of the safety of  
 314 the cylinder as a riser. Moreover, Figure 5 substantiates that the predictions of the proposed model in IL direction are close to  
 315 Trim's experimental results (Trim et al., 2005). The mean squared deviation of the displacements from the proposed model to  
 316 the experiment data is 0.11 for the IL direction and 1.26 for the CF direction.

317 The inaccuracy in predicting  $\xi$  is mainly attributed to the inaccuracy of the damping model, including the estimation of the

318 hydrodynamic damping coefficient  $\lambda$  and structural damping ratio  $\xi_D$ .  $\lambda$  usually takes the value of 0.8 as reported in the  
319 literature, but the simulation results suggest that the value of  $\lambda$  should be around 0.45 for the prediction of the proposed  
320 model to better match the experimental data.

321 In order to directly illustrate the cylinder vibrations simulated by the proposed model, the RMS of displacements in both IL  
322 and CF directions at  $U=1\text{m/s}$  are presented in Figures 6. It is apparent that the 13<sup>th</sup> and 7<sup>th</sup> modes are predominant for the IL  
323 and CF vibrations of the cylinder, respectively. Such findings are in line with the numerical and experimental investigations  
324 on the vibrations of the long flexible cylinders. In fact, the predominant mode number shown in Figures 6 indicates that the  
325 IL vibrations are with the frequency twice as large as the frequency of the CF vibrations. The same mode number is obtained  
326 as that from Gu et al. (2012) except the difference between the amplitudes. In addition, the IL vibrations are not included in  
327 the simulation conducted by Gu et al. (2012), and therefore not shown in Figure 6.

328 Figure 7 gives the time histories of non-dimensional displacements of  $y/D$  and  $x/D$  at different locations, with  $z/L$  equal to  
329 0.84, 0.67, 0.5, 0.33 and 0.16 at  $U=1\text{m/s}$ , and corresponding response spectra. It is evident that the displacement follows a  
330 precisely periodic trend and the segment at different elevations vibrates at the same frequency. With a specific case shown in  
331 Figure 7, the cylinder vibration frequencies, regardless of the location along its length, in CF and IL directions are 6.29Hz  
332 and 12.5Hz, respectively. As reported in Vandiver et al.'s study (Vandiver et al., 2009), there are two harmonic components  
333 in IL or CF vibrations and their intensities are different at different locations. More specifically, the harmonic component with  
334 higher frequencies weakens approaching to the end of the cylinder. The reason why only a harmonic vibration with a single  
335 frequency is produced in the numerical simulation reported in the present and similar investigations could be the variation in  
336 structural properties of the cylinder. More specifically, the experiment employs a cylinder that unavoidably contains flaws in  
337 the manufacturing, which results in variations in structural properties along the length of the cylinder.

338 Figure 8 and 9 show the trajectories of the vibration of cylinder segments at different elevations at  $U=1\text{m/s}$  and evolutions of  
339 the non-dimensional displacement at  $U=1.5\text{m/s}$ . In Figure 8, the trajectories shifted back and forth between the traditional

340 “figure 8” and the typical “crescent” shapes. It is argued that many factors could impact trajectory patterns, such as the cylinder  
341 scale, the ratio of IL natural frequency to CF natural frequency, mass ratio and the IL and CF frequencies (Kang et al., 2016).  
342 The trajectories shown in Figure 8 are the evidences that IL and CF waves are not phase locked. Such feature, i.e. the figure-  
343 of-eight trajectory, is also observed by Srinil and Zanganeh (2012b). In Figure 9, a traveling wave is observed as reported in  
344 the literature and its propagation direction is arbitrary (Violette et al., 2007). Compared to Ge et al.’s results (Ge et al., 2009),  
345 the predicted amplitudes of the proposed model are higher, and closer to Trim’s experimental results (Trim et al., 2005). In  
346 addition, the same dominant mode numbers for the vibrations in both IL and CF directions are observed. The similarities and  
347 differences in amplitude and dominant mode numbers are consistent with the findings shown in Figure 2 and 4 for the two  
348 numerical models. With the increase in flow velocities, dominant wave pattern shifted from standing wave to traveling wave.

## 349 **Conclusions**

350 A three-dimensional model predicting the VIV in both the IL and CF directions, coupled with a set of modified Van Der Pol  
351 equations, is presented. Fluid forces, including the lift and drag forces, due to the vortex-shedding are modelled by the flow  
352 variables the same as that in other semi-empirical models which contain a wake oscillator. A new tension formula is proposed  
353 to account for the variations in tensions due to the prolongations occurring in the cylinder. The proposed model (especially  
354 the tension formula) is validated by comparing to the available experimental data and numerical results. The comparison  
355 shows that the present model is capable of simulating VIV of long flexible cylinders in both the CF and IL directions under  
356 the influence of uniform incoming flow. Since the tension is modelled as a function of the stream velocity, the dynamics of  
357 cylinder VIV is more realistically simulated in this study. In fact, it is found that the proposed model outperforms the model  
358 proposed by Ge et al. (2009) and Gu et al. (2012) in predicting the variations for the vibration amplitudes and frequencies and  
359 dominant mode number with the stream velocity. Most importantly, this is the first attempt, to the best of the author’s  
360 knowledge, to propose a model containing dynamically determined tension to predict VIV in both the CF and IL directions.

361 Some aspects of long slender cylinders undergoing VIV can be reproduced qualitatively and quantitatively, such as dominant  
362 mode number, vibration frequency, amplitude and traveling wave phenomenon. Future research will be focused on the  
363 physical meaning of the model parameters.

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## APPENDIX: LIST OF SYMBOLS

459		
460	$M$	the sum of structure mass $m_s$ and added fluid mass per unit length
461	$F$	hydrodynamic force per unit length, with components $F_x$ and $F_y$
462	$r$	the deflection, a vector, with components $x$ and $y$
463	$EI$	bending stiffness
464	$E$	Young's modulus
465	$T$	axial tension
466	$C_a$	added mass coefficient
467	$\rho$	density of seawater
468	$D$	diameter of the riser
469	$R$	damping coefficient due to hydrodynamic damping and structural damping
470	$R_f$	damping coefficient due to hydrodynamic force
471	$R_s$	damping coefficient due to structure force
472	$\gamma$	parameter determined through experiments
473	$\Omega_f$	vortex shedding frequency
474	$\Omega_n$	natural frequency of the riser in air
475	$\Omega_s$	natural frequency of the riser in water
476	$\zeta_D$	damping ratio
477	$T_{ini}$	tension force before deflection
478	$A$	cross section area of the riser
479	$L$	length of the riser
480	$\Delta L$	prolongation of the riser

481	$F_x$	external force exerted perpendicularly on the model in the x direction
482	$F_y$	external force exerted perpendicularly on the model in the y direction
483	$f_D$	mean drag force per unit length
484	$f_D'$	fluating drag force per unit length
485	$f_L$	lift force per unit length
486	$\bar{C}_D$	total drag coefficient during vortex-shedding
487	$C_{D0}$	mean drag coefficient of a stationary rigid cylinder
488	$Y_{RMS/D}$	the amplitude of vibration in the CF direction
489	$C_{Di}$	vortex shedding drag coefficient
490	$C_{Di0}$	the amplitude of vortex shedding drag coefficient
491	$C_{L0}$	lift coefficient of a stationary rigid cylinder
492	$C_L$	lift coefficient
493	$p$	in line variable
494	$q$	cross flow variable
495	$\varepsilon_x, \varepsilon_y, A_x, A_y$	non-dimensional parameters estimated through experiments
496	$S_t$	Strouhal number