

Investigating the Effect of Residual Stress on Hydrogen Cracking in Multi-Pass Robotic Welding through Process Compatible Non-Destructive Testing

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Abstract

In this paper, the effect of Welding Residual Stress (WRS) on the size and morphology of hydrogen-induced cracks (HIC) is studied. Four samples were manufactured using a 6-axis welding robot and in two separate batches. The difference between the two batches was the clamping system used, which resulted in different amounts of welding deformation and WRS. The hydrogen cracks were intentionally manufactured in the samples using a localised water-quenching method, where water was sprayed over a specific weld pass in a predetermined position. The Phased-Array Ultrasonic Testing (PAUT) system was implemented during the welding process (high-temperature in-process method), to detect the HIC in real-time. The WRS in both batches was measured using the hole-drilling method, where a difference in transversal residual stress of 78MPa was found between the two samples. Based upon both the PAUT results and microscopic investigations, the batch with higher WRS resulted in larger size and number of HIC. For the first time, the negative effect of WRS on HIC has been monitored in real-time using high-temperature in-process inspection. This was achieved using an innovative approach, introduced in this paper, to repeatably manufacture high and low WRS samples in order to control the size and location of subsequent HIC.

Keywords: Multi-Pass Robotic Welding; Phased Array Ultrasonic Testing (PAUT); Hydrogen Induced Crack (HIC); Intentionally-Embedded Weld Defects; Welding Residual Stress (WRS); Hole-drilling Method.

31

32 **1. Introduction**

33 It is known that the initiation and propagation of Hydrogen Induced Cracks (HIC), also
34 known as cold-cracks, is influenced by three main factors: (I) hydrogen, (II) microstructure
35 and (III) residual stress [1]. Therefore, a higher amount of hydrogen penetrating into the weld
36 can increase the chance of HIC, especially in a brittle structure containing a higher amount of
37 residual stress [2]. Hydrogen induced cracking is considered a major contributor to the
38 increase in the repair costs associated with welding processes, e.g., £40 million of costs
39 incurred during manufacturing in the UK are due to the necessary repair of HIC [3]. Hence, it
40 is necessary to quantify the influencing parameters on the HIC initiation and propagation
41 using the inspection system.

42 In this study, the effect of residual stress on the HIC is investigated using a combination of
43 high-temperature in-process inspection using Phased-Array Ultrasonic Testing (PAUT),
44 robotic Non-Destructive Testing (NDT), Time-of-Flight Diffraction (TOFD) and hole-
45 drilling residual stress measurement. High-temperature in-process inspection of multi-pass
46 welding was discussed by Lines et al [4]. They used a flexible robotic cell for real-time NDT
47 of a weld sample which included some intentionally-embedded defects, i.e., tungsten rods [4,
48 5]. For PAUT, an ultrasonic array (64 elements) is used rather than a single element probe
49 since this method is expected to have higher resolution and a better signal to noise ratio [6,
50 7]. The offline inspection was also carried out using the PAUT system and TOFD, which
51 provides a 2D map of the defect positions in the weld length [8]. The resolution of PAUT
52 scans can be enhanced using a generic acquisition method called Full Matrix Capture (FMC)
53 combined with a post-processing technique called Total Focusing Method (TFM). This
54 allows to potentially focus on all points influenced by the ultrasonic wave [9].

55 It is believed that the welding process leads to the development of significant residual
56 stresses, the stress remaining in the component in the absence of any thermal gradient or
57 external forces [10]. The Welding Residual Stress (WRS) can be measured by destructive
58 methods (e.g., incremental deep hole drilling [11, 12] or the contour method [13, 14]) or by
59 NDT methods (e.g., ultrasonic [15-17], X-ray or neutron diffraction [11]). Among over 10
60 different residual stress measurement methods [10, 18], hole-drilling is the only method
61 which is standardised by ASTM E837 and, as such, is usually used for verification of the
62 other residual stress measurement methods [18-20]. Therefore, the hole-drilling method was

63 chosen for use in this study to measure the WRS in the samples with intentionally-
64 manufactured HIC.

65 In this paper, a high-temperature in-process monitoring system was used to detect the HIC
66 intentionally manufactured during the welding process. The inspection system included a
67 robotic arm equipped with a PAUT end-effector suitable for use in temperatures up to [150°
68 C]. Other non-contact NDT techniques such as thermography [21], electromagnetic acoustic
69 transducer (EMAT) [22], Laser-Induced Ultrasonic Phased Array (LIPA) [23], eddy current
70 [24] and X-ray/radiography [25], are potential alternative options for real-time inspection of
71 welds. However, due to the lower penetration depth of eddy current (potentially a few
72 millimetres [24]), the safety concerns surrounding radiography inspection [26], and the lower
73 resolution of thermography [24], less matured technology of LIPA [23] and lower signal-to-
74 noise ratio of EMAT [22, 24] when compared with the phased array ultrasonic method [24],
75 PAUT is preferred for this work.

76

77 **2. Experimental setup**

78 **2.1. Samples with intentionally-manufactured hydrogen cracks**

79 The four samples were manufactured from S275 structural steel plates (see Table 1 for the
80 chemical composition) with a thickness of 15 mm and a length of 300 mm. The multi-pass
81 weld included 21 passes, deposited in 7 layers, inside a 90° degree V-groove and was
82 performed using a Tungsten Inert Gas (TIG) welding process. All four samples were
83 manufactured with the same welding parameters, listed in Table 2, and layout, shown in
84 Figure 1. An Automatic Voltage Correction (AVC) system was used to keep the welding
85 voltage consistent throughout with real-time communication between the robot controller and
86 the welding machine facilitated KUKA Robot Sensor Interface (RSI) [27] whereupon the
87 welding voltage was adjusted with a continuously varied robot Z-position to control the arc
88 length.

89 **Table 1. Chemical composition of S275 structural steel (based on the material certificate provided by the**
90 **manufacturer)**

C	Si	Mn	P	S	Ni	Cr	Mo	Cu	V	Fe
0.12%	0.16%	0.57%	0.027%	0.023%	0.16%	0.181%	0.033%	0.55%	0.001%	Balance

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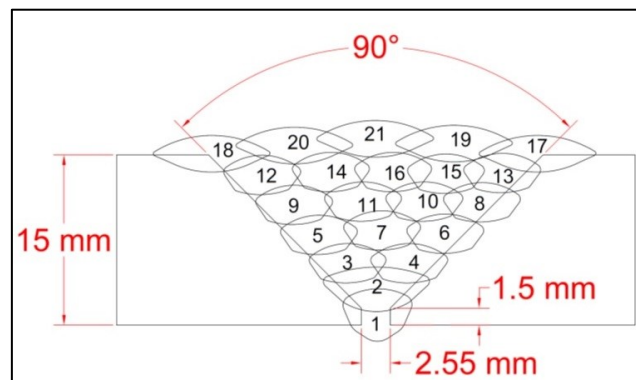
Table 2. Welding parameters

	AVC* set voltage (V)	Current (A)	Travel Speed (mm/min)	Wire Feed Speed (mm/min)	Weaving Amplitude (mm)	Weaving Frequency (Hz)	Inter-pass Temperature (° C)
Pass 1 (root pass)	12	120	50	910	2	0.3	80-100
Pass 2 (hot pass)	13.5	220	100	1225	4	0.6	80-100
Pass 3-16 (filling passes)	13.5	210	120	1470	3	0.55	80-100**
Pass 17-21 (capping passes)	13.5	240	100	1225	4	0.6	80-100

* Automatic Voltage Correction (AVC) using the RSI.

** Inter-pass temperature depends on the inspection time (as the subsequent weld pass starts immediately after the inspection of the last position) with an exception of the Pass 9 after which the localised water-quenching took a few minutes leading to much lower inter-pass temperature (near the room temperature especially over the quenched area).

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Figure 1. Welding layout and pass sequence used in this study

97

The welding wire was intentionally selected to be a hard material (i.e., hard-facing wire) to increase the probability of forming a HIC in the pre-determined position. This is a high carbon wire (0.5% C, 3% Si, 0.5% Mn, 9.5% Cr and Fe: balance) with a carbon equivalent (CE) higher than 0.4, meaning that it is more likely to form a martensitic brittle weld which will be prone to hydrogen cracking, especially with the existence of high WRS [28].

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A localised water quenching process was carried out directly after the deposition of Pass 9 in order to introduce hydrogen into the weld site. The process is known as *localised*-quenching since water is sprayed only after a specific pass and within only a small (40 mm) section of

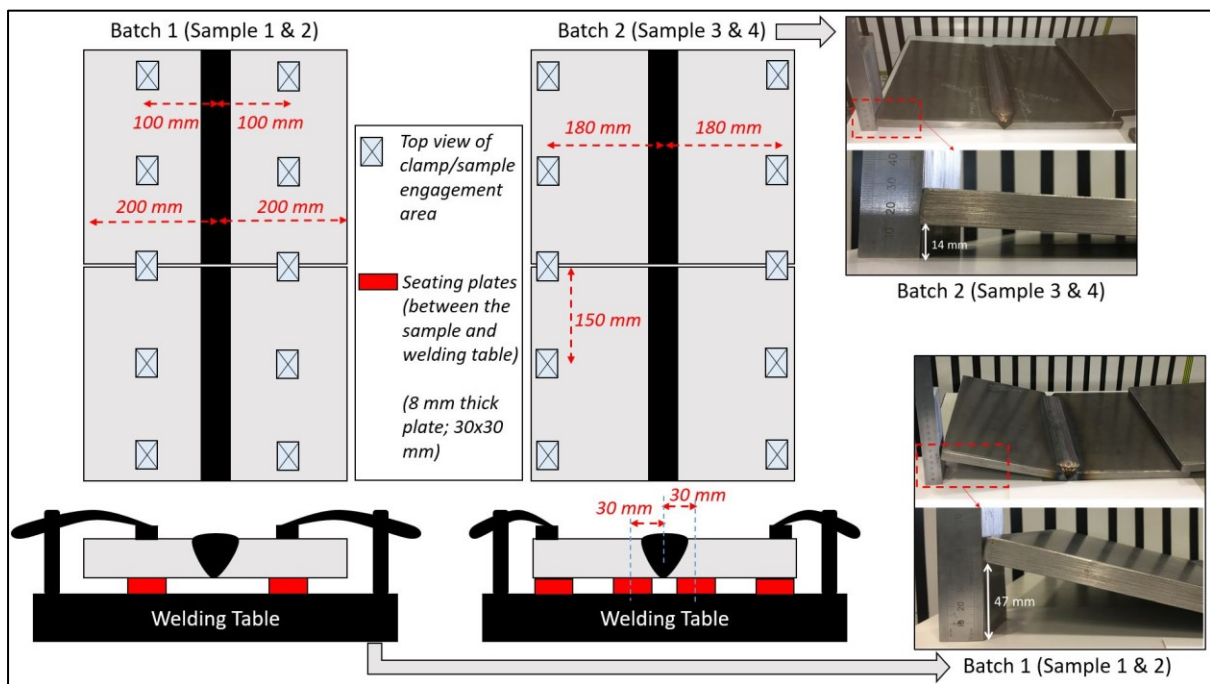
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105 the weld length (rather than over the full length of the weld). Water absorbers were used to
106 limit the quenching area and avoid any water spreading to other sections of the weld.

107 2.2. Manufacturing of high-stress and low-stress samples

108 Four specimens were manufactured in two batches as shown in Figure 2. Javadi et al [19]
109 showed that the position of clamps during the welding of stainless steel plates can influence
110 the amount of residual stress and deformation. Therefore, Batch 1 & 2 were manufactured
111 with different positioning of the clamps and seating plates as shown in Figure 2. This resulted
112 in a considerable difference in the angular shrinkage (47 mm in Batch 1 against only 14 mm
113 in Batch 2) which is due to the difference in the degree of freedom between two batches.
114 During the welding of Batch 1, the weld area could move down (due to the lack of seating
115 plates next to the weld) while the corners could move up (due to the lack of clamps in the
116 corners). These factors cause extensive angular shrinkage. It can be expected that there will
117 be lower residual stresses present in Batch 1 (since some WRS can be released in the form of
118 plastic deformation [10, 17, 19, 29]) when compared with Batch 2 where the deformation was
119 restricted by tight clamping and effective seating plate placement, see Figure 2. However, this
120 expected difference in the WRS must be proven using the residual stress measurement which
121 was carried out by the hole-drilling method in this study.



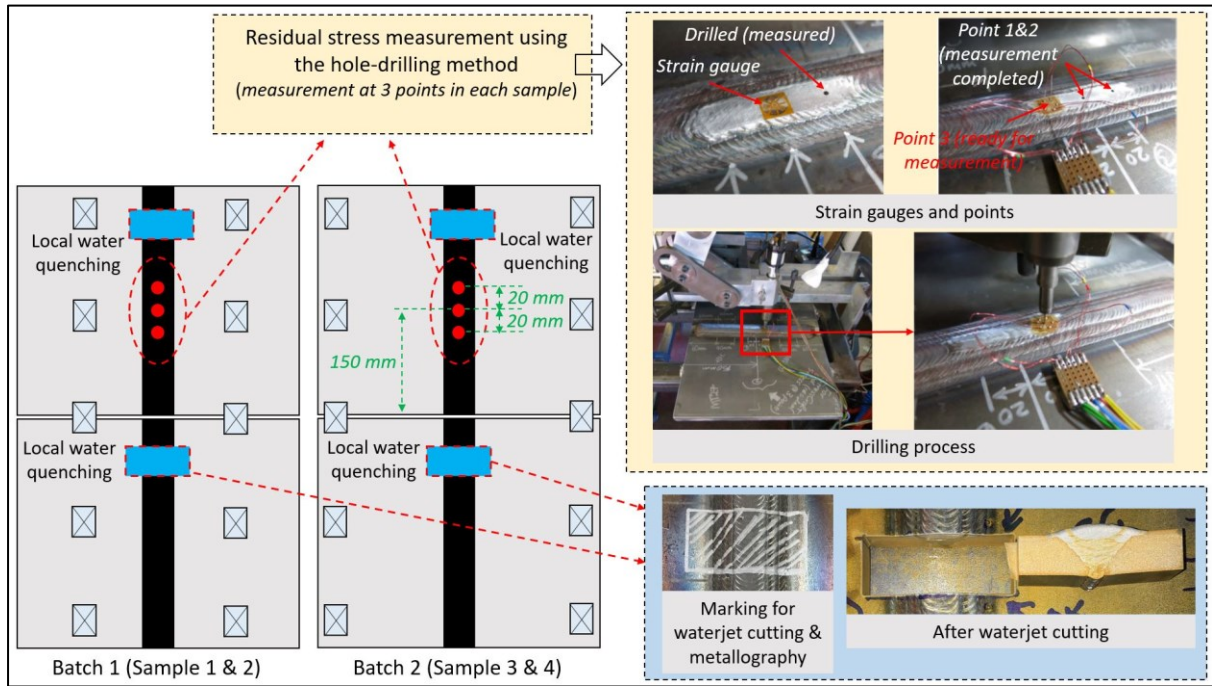
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123 **Figure 2. Welding and clamping design to manufacture high-stress and low-stress samples**

124

125 **2.3. Hole-drilling stress measurement and microscopic investigations**

126 Since both samples in each batch were manufactured using the same process parameters, and
127 the welding was carried out through a repeatable, fully robotic system, no difference between
128 samples from the same batch was anticipated. Therefore, one sample from each batch was
129 sent for residual stress measurement using the hole-drilling method and the remaining
130 samples were sent for metallography (see Figure 3).



132 **Figure 3. Position and setup for the residual stress measurement and metallography**

133 The hole-drilling process [30] was carried out based on ASTM E837, in three positions on
134 each sample. The first point is taken to be central with respect to both the weld length and
135 width while two further points are taken in order to increase the accuracy of the overall
136 measurement. The distance between the points is 20 mm to avoid the surface preparation
137 effect of one point interfering with the other points. The hole-drilling procedure includes
138 drilling a hole incrementally at each point, a strain gauge bonded to the surface measures the
139 strain at each increment [31]. These strains can be related to the amount of residual stress
140 which is released in the form of deformation [31]. The material properties of the weld,
141 deposited using the hard-facing wire, were as follows: Young's modulus of 205 GPa and
142 Poisson's ratio (μ) of 0.295. After surface preparation, the strain gauges were bonded such
143 that each gauge was orientated with element 1 in the weld direction and element 3 in the
144 transverse direction (perpendicular to the weld). Holes were drilled at 16 depth increments set

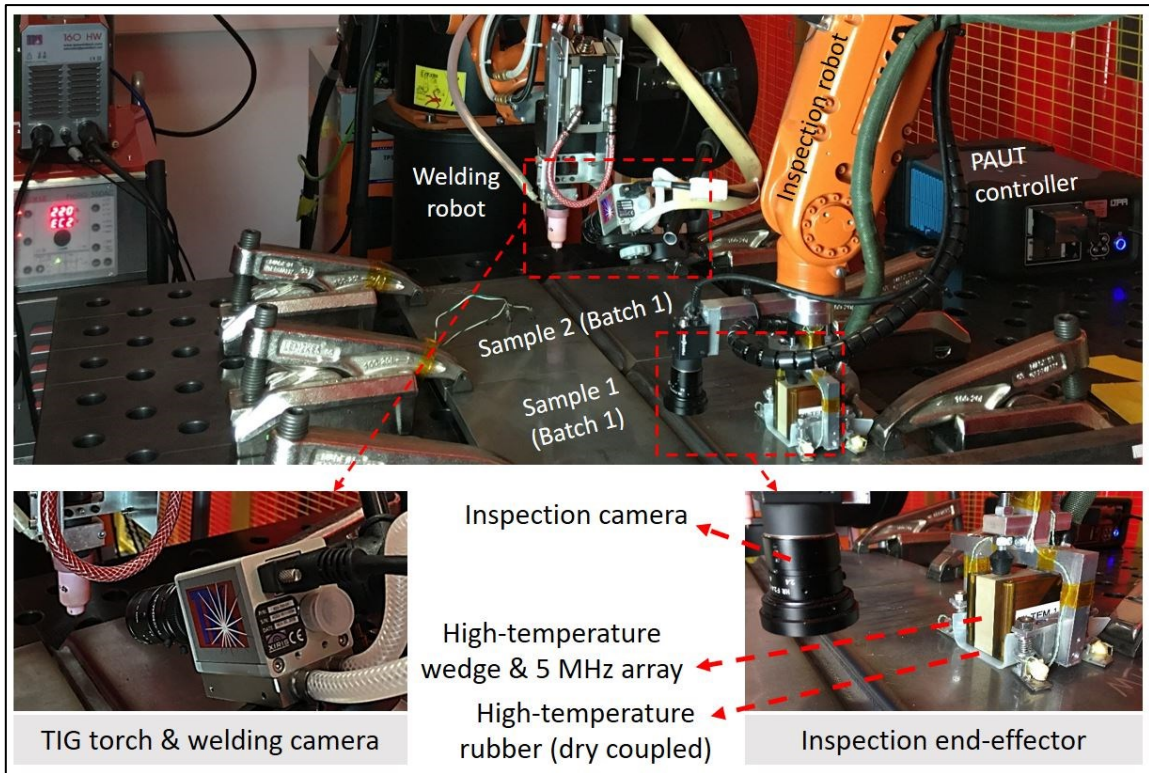
145 at 4 x 32 μm + 4 x 64 μm + 8 x 128 μm to give a completed hole depth of 1.4 mm. All gauge
146 mounting and drilling procedures were conducted in accordance with the National Physical
147 Laboratory good practice [31, 32].

148 The remaining sample from each batch was marked for cutting, based on off-line ultrasonic
149 inspection results, within the local-quenching area where the HIC is expected to be observed
150 clearly (see Figure 3). The marked area was first removed using a water-jet cutting process,
151 however, due to the high hardness of the weld material, extra surface preparation (milling
152 with carbide tools and grinding) was necessary before the metallography. The samples were
153 then etched for 20 seconds using a mixture of nitric acid (one part) and ionised water (three
154 parts).

155

156 ***2.4. Robotic welding and high-temperature in-process inspection system***

157 Both Batch 1 and 2 were manufactured using a robotic welding process in which a TIG
158 source was used (see Figure 4). The welding process was monitored using a high dynamic
159 range camera. The inspection process was also carried out by a 6-axis KUKA robot to
160 implement PAUT and visual inspection of the weld surface as shown in Figure 4. The
161 inspection end-effector included a 5 MHz ultrasonic array (64 elements) mounted on a high-
162 temperature wedge (Olympus ULTEM wedge) and an inspection camera. Although a high-
163 temperature gel-couplant (Olympus high-temperature couplant) was used between the wedge
164 and rubber, no couplant was used between the rubber and the specimen surface. This dry-
165 couplant inspection technique was critical to avoid any couplant contaminating the weld site
166 unintentionally and causing uncontrollable weld-defects. Since the inspection end-effector
167 was equipped with high-temperature devices (high-temperature wedge, couplant and rubber),
168 it was possible to carry out the inspection process between the deposition of the welding
169 passes (i.e., when the specimen surface temperature is $<150\text{ }^{\circ}\text{C}$). This allowed for real-time
170 PAUT sector-scanning of the hydrogen crack initiation and growth.



171

172

Figure 4. Robotic welding and high-temperature in-process inspection system

173

174 **2.5. Offline inspection (PAUT and TOFD)**

175 The samples were tested using PAUT (5 MHz array, high-temperature wedge and couplant)
 176 and TOFD as shown in Figure 5. The PAUT imaging approach included both sector scanning
 177 and also post-processing, i.e., TFM. The accuracy of this offline process was critical in order
 178 to exactly mark the samples for metallography because the small cracks and subsequent
 179 narrow defect area could be accidentally removed during the heavy machining required for
 180 surface preparation of this hard-facing wire.

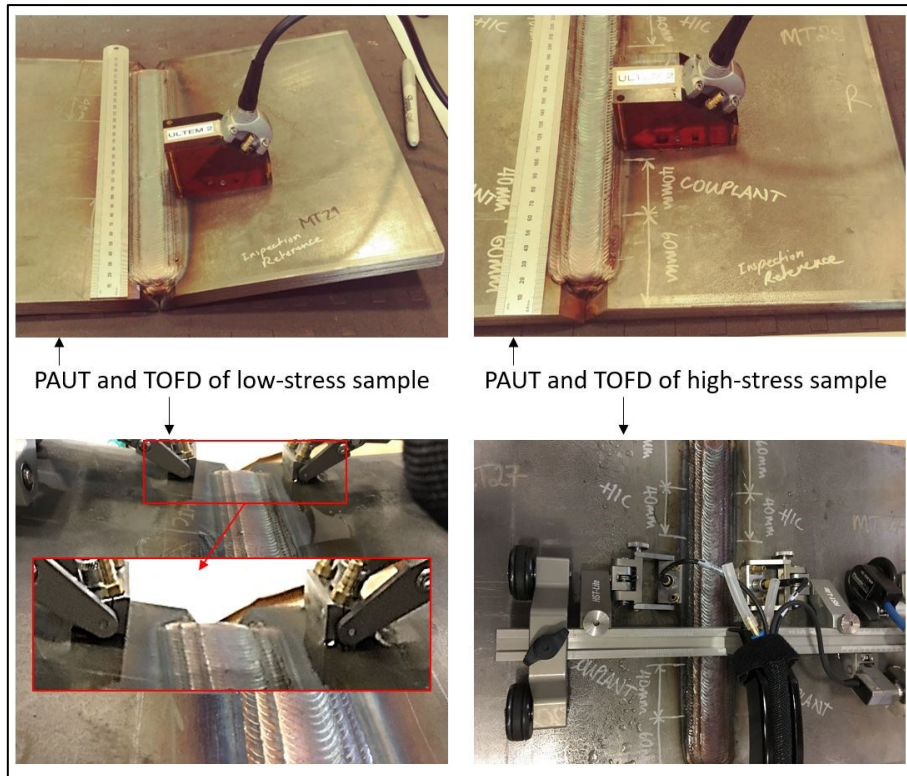


Figure 5. Inspection using PAUT (sector scanning) and TOFD

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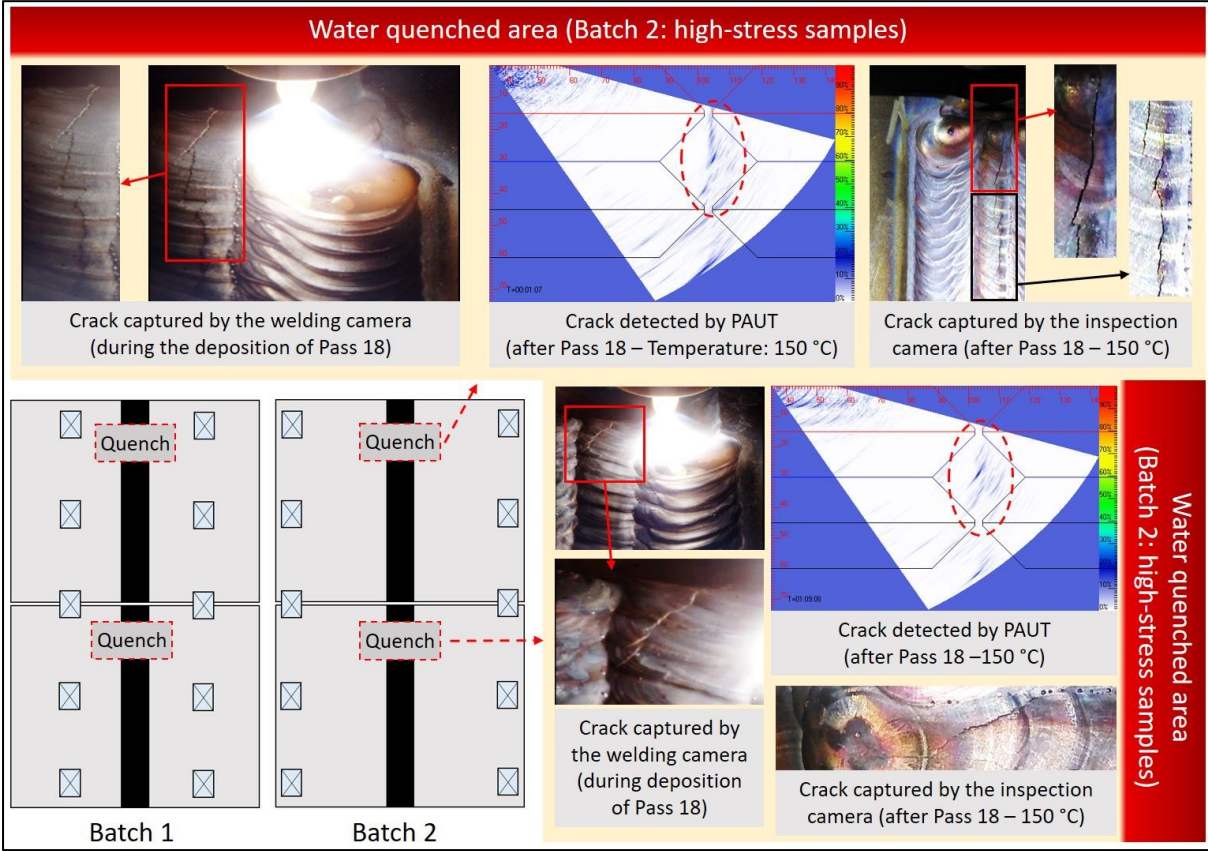
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184 3. Results and discussions

185 3.1. High-temperature in-process inspection results

186 Each of the four samples manufactured in this work was inspected in three inspection
 187 positions after each of the 21 welding-passes. The inspection positions were selected in a way
 188 that covered both the localised water-quenching area and areas which were considered defect-
 189 free. There were some obvious reflectors detected by the high-temperature PAUT system and
 190 they were captured by the welding and inspection cameras as well (see Figure 6). It is worth
 191 mentioning that these cracks were only observed in Batch 2 (high-stress samples) while none
 192 of the three inspection devices (PAUT, welding camera and inspection camera) detected any
 193 obvious cracks in Batch 1 (low-stress samples). However, this does not mean that there were
 194 no cracks manufactured in Batch 1 since two of three inspection devices (i.e., welding and
 195 inspection camera) can only capture surface cracks. Therefore, it is possible that some
 196 internal cracks were produced in Batch 1 but were small enough in size that the high-
 197 temperature PAUT resolution was not high enough to detect them. This can be proven either

198 by the metallography of the samples or by the offline PAUT at room temperature, where
 199 higher resolution is expected.



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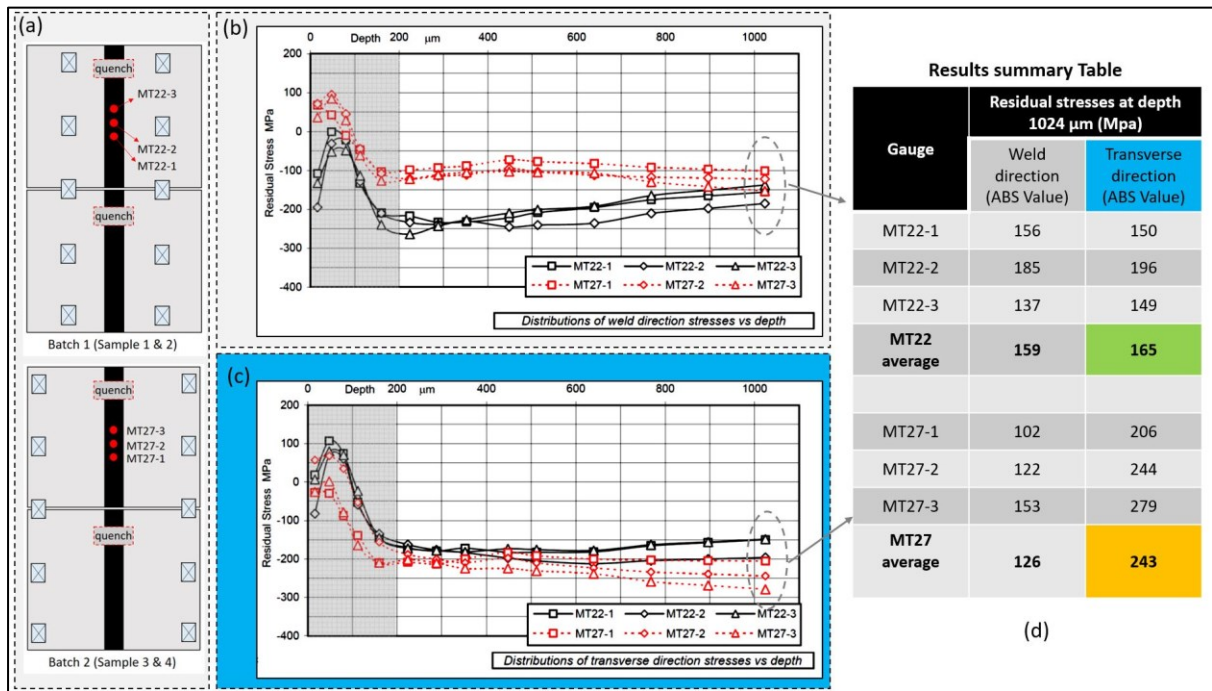
201 **Figure 6. High-temperature in-process inspection of the crack in Batch 2 (high-stress samples)**

202

203 **3.2. Residual stress measurements**

204 The results of residual stress measurement using the hole-drilling method are shown in Figure
 205 7. It is worth noting that one sample with potentially lower stress (i.e., Batch 1), MT22 based
 206 on the annotation system of the lab hosting this work, and another with potentially higher
 207 stress (Batch 2: MT27) was sent for stress measurement. The NPL Good Practice Guide [31]
 208 lists a number of contributors to stress uncertainty, including factors arising from the
 209 component, the drilling process, the strain gauge and strain indicator. The strain gauge and
 210 indicator together are the greatest sources of uncertainty in the form of noise in the strain
 211 output. A random strain uncertainty in the range $\pm 3 \mu\epsilon$ applied to the strain data of gauges
 212 used in this assessment produces uncertainties of $\sim \pm 60$ MPa in near-surface σ_1 and σ_3
 213 stresses. The high level of uncertainties near the surface is likely due to the practical
 214 difficulties of surface preparation as a result of the high hardness associated with the hard-

215 facing wire. This uncertainty decreases to a minimum of ± 11 MPa at depth 512 μm and then
 216 increases again to ± 24 MPa at the final increment due to sensitivity reduction. Therefore, the
 217 near-surface defects are shown in a grey area in Figure 7b,c as they are not considered in this
 218 paper due to the high level of uncertainty.



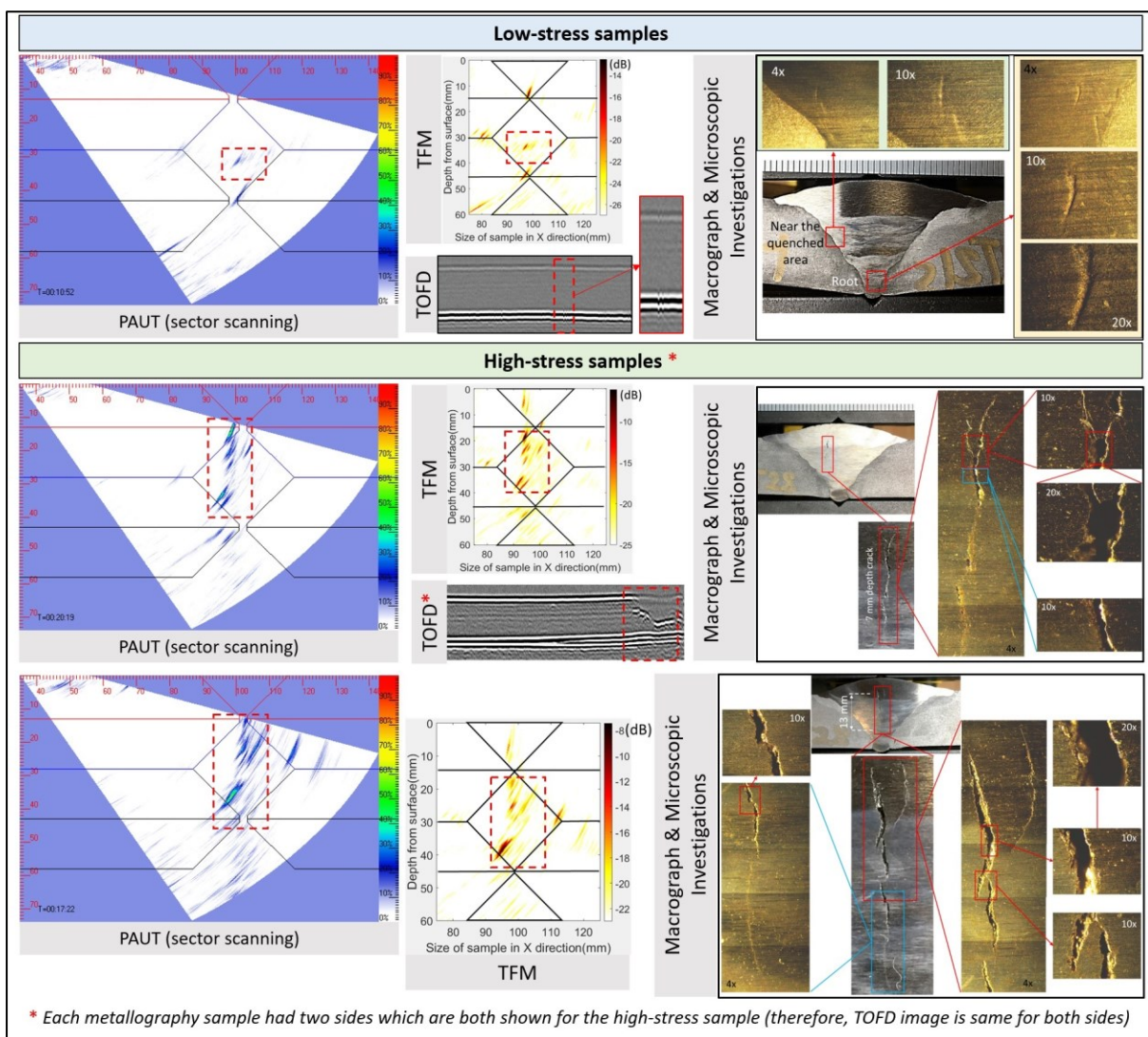
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 220 **Figure 7. Residual stress measurement results (a: the measurement positions – b: residual stress in the**
 221 **weld direction, longitudinal stress – c: residual stress in the transverse direction, transversal stress – d:**
 222 **results summary table)**

223
 224 The average results of residual stress measured at the last increment of the hole-drilling
 225 method (1024 μm) are highlighted in a summary table in Figure 7d. These results are
 226 believed to be the most relevant to the HIC since they are taken from the closest possible
 227 point to the quenched area. The cracks that were captured by the inspection camera were
 228 longitudinal in nature (see Figure 6) and therefore, the transversal residual stresses had the
 229 main effect on the direction of crack propagation. Hence, the average value of transversal
 230 residual stresses is highlighted in the summary table in Figure 7d. Results show that Batch 1
 231 (MT22) residual stress reaches to the absolute value of 165 MPa which is considerably lower
 232 than Batch 2 residual stress, MT27 with 243 MPa. This proves the main idea of this paper in
 233 which Batch 1 had been designed in a way to result in lower stress than Batch 2. Hence, it is
 234 expected to detect smaller hydrogen cracks in Batch 1 in comparison with Batch 2.

235

236 **3.3. PAUT, TOFD and microscopic investigation results**

237 The PAUT (sector scanning), TFM, TOFD, macrograph and microscopic investigation results
 238 are shown in Figure 8. It shows an agreement between all results, i.e., stronger PAUT and
 239 TFM signal are matched with stronger TOFD reflection and both are in agreement with the
 240 microscopic investigations. Furthermore, all of these results are in line with the main idea of
 241 this paper - proving that a large crack was detected in the sample on which higher residual
 242 stress was measured. It is worth mentioning that the hydrogen crack was expected to initiate
 243 from the area where the samples were quenched. This has been clearly observed in all of the
 244 cracks detected in the high-stress samples.



245

246 **Figure 8. PAUT (sector scanning), TFM, TOFD and microscopic investigation results**

247

248 Regarding the low-stress samples, there are some signs of cracking. This is evident in the
249 inspection results (PAUT, TOFD and TFM) where a weak reflection signal is detected (see
250 Figure 8). The macrographs also show some cracks near the quenched area and in the root
251 pass. These cracks are believed to be hydrogen cracks since they are propagated only in the
252 40 mm length of the water-quenched area while the rest of the weld length was defect-free, as
253 proven by the TOFD inspection results (see Figure 8).

254 Finally, a very obvious difference in the length and size of HIC in the low and high-stress
255 samples can be concluded. This was reflected in the offline inspection results (PAUT, TFM
256 and TOFD signals) and microscopic investigations, shown in Figure 8, along with the in-
257 process inspections (visual camera and high-temperature PAUT system) discussed in Sec.
258 3.1.

259

260 **4. Conclusions**

261 In this paper, a combination of phased array ultrasonic testing, TOFD, TFM, microscopic
262 investigation, residual stress measurement, high-temperature and in-process robotic NDT was
263 used to study the effect of welding residual stress on intentionally-manufactured hydrogen
264 cracks. Based on the results, it can be concluded that:

- 265 1) The high-temperature in-process inspection system (PAUT, inspection and welding
266 camera) detected a number of large cracks during the deposition of filling passes of
267 the multi-pass welding process. However, the cracks were only visible during the
268 welding of high-stress samples, while the low-stress samples did not show any signs
269 of in-process cracking.
- 270 2) The residual stress measurement using the hole-drilling method proved that the
271 clamping design was successful for manufacturing samples with high and low
272 residual stress values (a difference of 78 MPa was measured in the transverse
273 direction).
- 274 3) The macrograph and microscopic investigations showed a number of very large
275 hydrogen cracks had been intentionally manufactured in the quenched area of the
276 high-stress sample. There are also some traces of cracks in the low-stress samples, but
277 they are considerably smaller than the cracks propagated in the high-stress sample.

278 4) The PAUT, TOFD and TFM results were in good agreement with the macrographs
279 and microscopic investigations. Therefore, larger cracks that were subsequently
280 detected in the macrograph had already been detected with a stronger signal in both
281 PAUT and TOFD inspections. This was also in good agreement with the residual
282 stress measurement results as the larger cracks (and stronger signals) were detected in
283 the high-stress samples.

284 Therefore, the combination of inspection systems developed in this paper has been shown
285 to successfully detect the negative effect which residual stress has on the structural integrity
286 of the weld components through the development of larger hydrogen cracks. For the
287 first time, this negative effect of WRS on HIC and the relationship between the two has
288 been monitored in real-time using high-temperature in-process inspection. Furthermore,
289 the achievements of this paper would not have been possible without the innovative
290 approach developed to repeatably manufacture test samples with control over the WRS and
291 size and location of HIC.

292

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297 Inspection”). Furthermore, the concept of high-temperature in-process inspection will be used
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304

305 **6. References**

306 [1] Lippold JC. Hydrogen-Induced Cracking. *Welding Metallurgy and Weldability* 2014.

307 [2] Barrera O, Bombac D, Chen Y, Daff TD, Galindo-Nava E, Gong P, et al. Understanding
308 and mitigating hydrogen embrittlement of steels: a review of experimental, modelling and
309 design progress from atomistic to continuum. *Journal of Materials Science*. 2018;53:6251-90.
310 [3] Bailey N, Coe FR, Gooch TG, Hart PHM, Jenkins N, Pargeter RJ. Chapter 1 - Defining
311 the problem. In: Bailey N, Coe FR, Gooch TG, Hart PHM, Jenkins N, Pargeter RJ, editors.
312 *Welding Steels Without Hydrogen Cracking (Second Edition)*: Woodhead Publishing; 2004.
313 p. 1-16.
314 [4] Lines DI, Javadi Y, Mohseni E, Vasilev M, MacLeod CN, Vithanage RW, et al. Flexible
315 robotic cell for in-process inspection of multi-pass welds. *NDT2019: 58th Annual*
316 *Conference of the British Institute of NDT*. Telford, United Kingdom 2019.
317 [5] Javadi Y, MacLeod CN, Pierce SG, Gachagan A, Lines D, Mineo C, et al. Ultrasonic
318 phased array inspection of a Wire + Arc Additive Manufactured (WAAM) sample with
319 intentionally embedded defects. *Additive Manufacturing*. 2019;29:100806.
320 [6] Drinkwater BW, Wilcox PD. Ultrasonic arrays for non-destructive evaluation: A review.
321 *Ndt & E International*. 2006;39:525-41.
322 [7] Javadi Y, MacLeod CN, Pierce SG, Gachagan A, Kerr W, Ding J, et al. Ultrasonic phased
323 array inspection of wire plus arc additive manufacture (WAAM) samples using conventional
324 and total focusing method (TFM) imaging approaches. *Insight*. 2019;61:144-8.
325 [8] Yeh FWT, Lukomski T, Haag J, Clarke T, Stepinski T, Strohaecker TR. An alternative
326 Ultrasonic Time-of-Flight Diffraction (TOFD) method. *Ndt & E International*. 2018;100:74-
327 83.
328 [9] Holmes C, Drinkwater BW, Wilcox PD. Post-processing of the full matrix of ultrasonic
329 transmit-receive array data for non-destructive evaluation. *Ndt & E International*.
330 2005;38:701-11.
331 [10] Withers PJ, Turski M, Edwards L, Bouchard PJ, Buttle DJ. Recent advances in residual
332 stress measurement. *International Journal of Pressure Vessels and Piping*. 2008;85:118-27.
333 [11] Javadi Y, Smith MC, Abburi Venkata K, Naveed N, Forsey AN, Francis JA, et al.
334 Residual stress measurement round robin on an electron beam welded joint between
335 austenitic stainless steel 316L(N) and ferritic steel P91. *International Journal of Pressure*
336 *Vessels and Piping*. 2017;154:41-57.
337 [12] Mahmoudi AH, Hossain S, Truman CE, Smith DJ, Pavier MJ. A New Procedure to
338 Measure Near Yield Residual Stresses Using the Deep Hole Drilling Technique.
339 *Experimental Mechanics*. 2009;49:595-604.
340 [13] Javadi Y, Walsh JN, Elrefaey A, Roy MJ, Francis JA. Measurement of residual stresses
341 induced by sequential weld buttering and cladding operations involving a 2.25Cr-1Mo
342 substrate material. *International Journal of Pressure Vessels and Piping*. 2017;154:58-74.
343 [14] Prime MB, Hill MR, DeWald AT, Sebring RJ, Dave VR, Cola MJ. Residual stress
344 mapping in welds using the contour method. *Trends in Welding Research, Proceedings*.
345 2003:891-6.
346 [15] Javadi Y, Ashoori M. Sub-surface stress measurement of cross welds in a dissimilar
347 welded pressure vessel. *Materials & Design*. 2015;85:82-90.
348 [16] Javadi Y, Akhlaghi M, Najafabadi MA. Using finite element and ultrasonic method to
349 evaluate welding longitudinal residual stress through the thickness in austenitic stainless steel
350 plates. *Materials & Design*. 2013;45:628-42.
351 [17] Javadi Y. Investigation of Clamping Effect on the Welding Residual Stress and
352 Deformation of Monel Plates by Using the Ultrasonic Stress Measurement and Finite
353 Element Method. *Journal of Pressure Vessel Technology-Transactions of the Asme*.
354 2015;137.
355 [18] Rossini NS, Dassisti M, Benyounis KY, Olabi AG. Methods of measuring residual
356 stresses in components. *Materials & Design*. 2012;35:572-88.

357 [19] Javadi Y, Hasani M, Sadeghi S. Investigation of Clamping Effect on the Welding Sub-
358 surface Residual Stress and Deformation by Using the Ultrasonic Stress Measurement and
359 Finite Element Method. *Journal of Nondestructive Evaluation*. 2015;34.

360 [20] Javadi Y, Akhlaghi M, Najafabadi MA. Nondestructive Evaluation of Welding Residual
361 Stresses in Austenitic Stainless Steel Plates. *Research in Nondestructive Evaluation*.
362 2014;25:30-43.

363 [21] Vasudevan M, Chandrasekhar N, Maduraimuthu V, Bhaduri AK, Raj B. Real-Time
364 Monitoring of Weld Pool during GTAW using Infra-Red Thermography and analysis of
365 Infra-Red thermal images. *Welding in the World*. 2011;55:83-9.

366 [22] Latham WM, MacLauchlan DT, Geier DP, Lang DD. EMAT weld inspection and weld
367 machine diagnostic system for continuous coil processing lines: SPIE; 1996.

368 [23] Stratoudaki T, Javadi Y, Kerr W, Wilcox PD, Pieris D, Clark M. Laser Induced Phased
369 Arrays for remote ultrasonic imaging of additive manufactured components. 57th The British
370 Institute of Non-destructive Testing Annual Conference. Nottingham, United Kingdom: The
371 British Institute of Non-destructive Testing (BINDT); 2018. p. 10.

372 [24] Shull PJ. *Nondestructive evaluation: theory, techniques, and applications*: CRC press;
373 2002.

374 [25] Sun Y, Bai P, Sun HY, Zhou P. Real-time automatic detection of weld defects in steel
375 pipe. *Ndt & E International*. 2005;38:522-8.

376 [26] Cawley P. Non-destructive testing - current capabilities and future directions.
377 *Proceedings of the Institution of Mechanical Engineers Part L-Journal of Materials-Design
378 and Applications*. 2001;215:213-23.

379 [27] KUKA. *KUKA RobotSensorInterface (RSI) 3.2 Documentation* . In: KUKA, editor.
380 Version: KST RSI 32 V12013.

381 [28] Vergani L, Colombo C, Gobbi G, Bolzoni FM, Fumagalli G. Hydrogen Effect on
382 Fatigue Behavior of a Quenched&tempered Steel. *Procedia Engineering*. 2014;74:468-71.

383 [29] Francis JA, Bhadeshia HKDH, Withers PJ. Welding residual stresses in ferritic power
384 plant steels. *Materials Science and Technology*. 2007;23:1009-20.

385 [30] Schajer GS, Whitehead PS. Hole Drilling and Ring Coring. In: Schajer GS, editor.
386 *Practical Residual Stress Measurement Methods*: BLACKWELL SCIENCE PUBL,
387 OXFORD, ENGLAND; 2013.

388 [31] Grant PV, Lord JD, Whitehead PS. *National Physical Laboratory Good Practice Guide
389 No. 53 Issue 2. The measurement of Residual Stresses by the Incremental Hole Drilling
390 Technique*. UK: National Physical Laboratory; 2006.

391 [32] Whitehead PS. *Stresscraft Ltd; Methods and Techniques. The determination of residual
392 stresses using the target gauge/centre hole drilling technique*. UK: Stresscraft Ltd; 2018.

393